

European SpaceCraft for the study of Atmospheric Particle Escape

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and ESCAPE proposal team*

A mission proposal to **ESA's M5 call** to understand the atmospheric escape from the Earth, a magnetized planet.

Why atmospheric escape?

Compositions of the Earth, Mars, and Venus are so different

(e.g., D/H ratio indicates water at Mars/Venus was 0.1% of the Earth 4.5 Byr ago, and Thermal (inc. photochemical) mechanism cannot explain water)

⇒ **Estimate contributions from different escape mechanisms**
with different mass-filtering efficiency

How much is the Earth special in terms of habitability?

(Evolution of the atmospheric composition and ocean pH must have allowed life to survive and evolve. If the escape was too much or too little, such condition might not be satisfied. Inversely, B-field, biologic & subsurface activities played roles?)

⇒ **Estimate possible roles of escape in habitability**

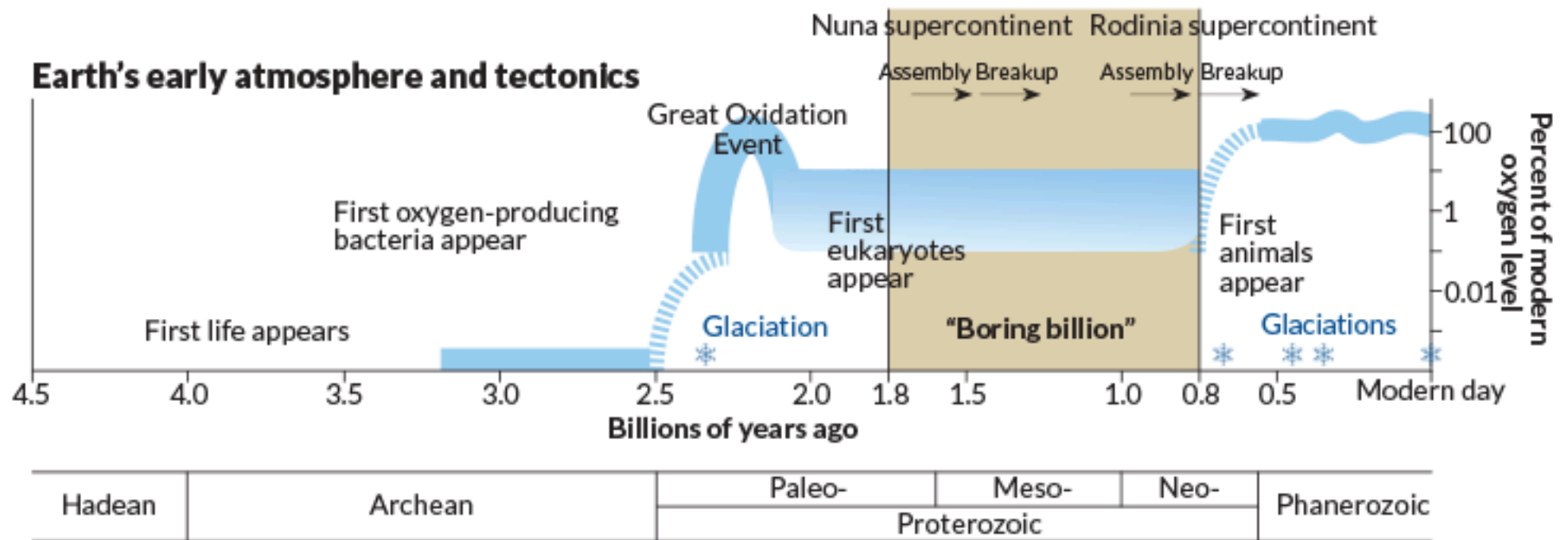
Life might have extincted and revived many times???

(Atmospheric conditions during the life emergence is one of the key)

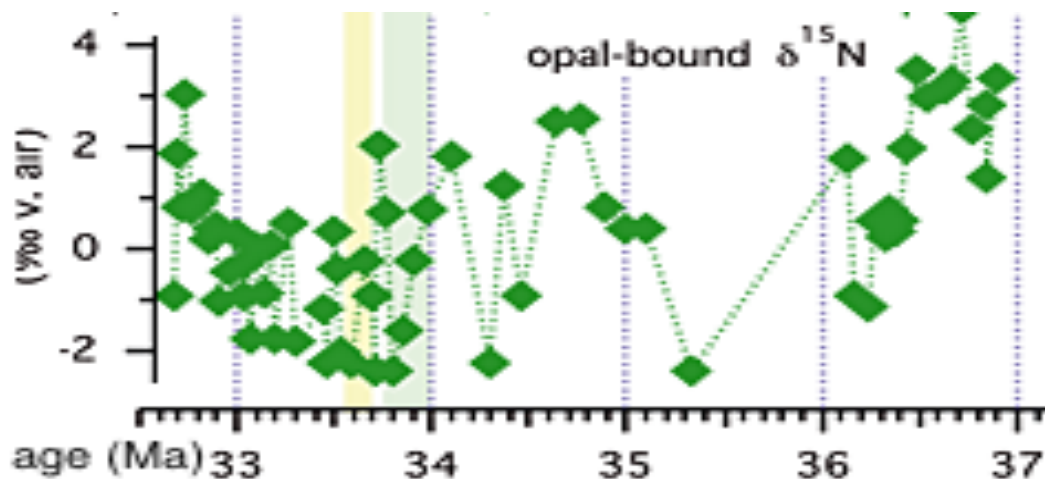
⇒ **Estimate atmospheric evolution right after life emerged**

⇒ **Need independent studies from subsurface route**

(Subsurface inventory is large, but could recycle without loss)



O₂ model (*Lyons et al., 2014*)



Ocean ¹⁵N (*Robinson et al., 2015*)

Past N/O ratio fluctuated a lot

How much ion escape explains N/O ratio & pH changes?

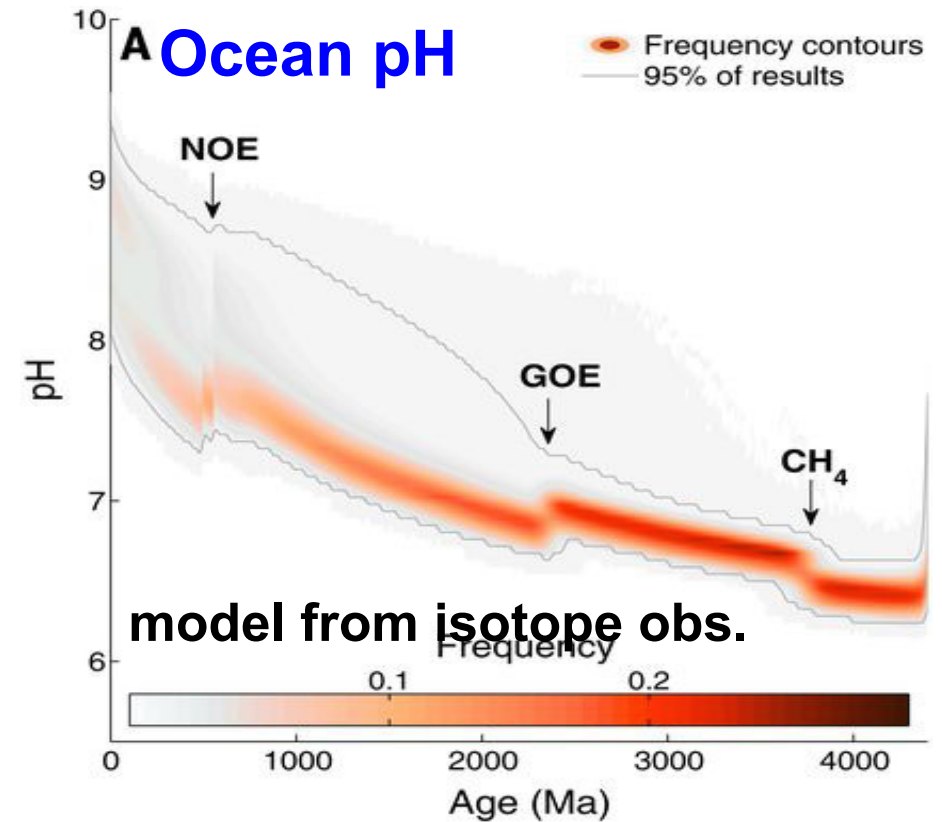
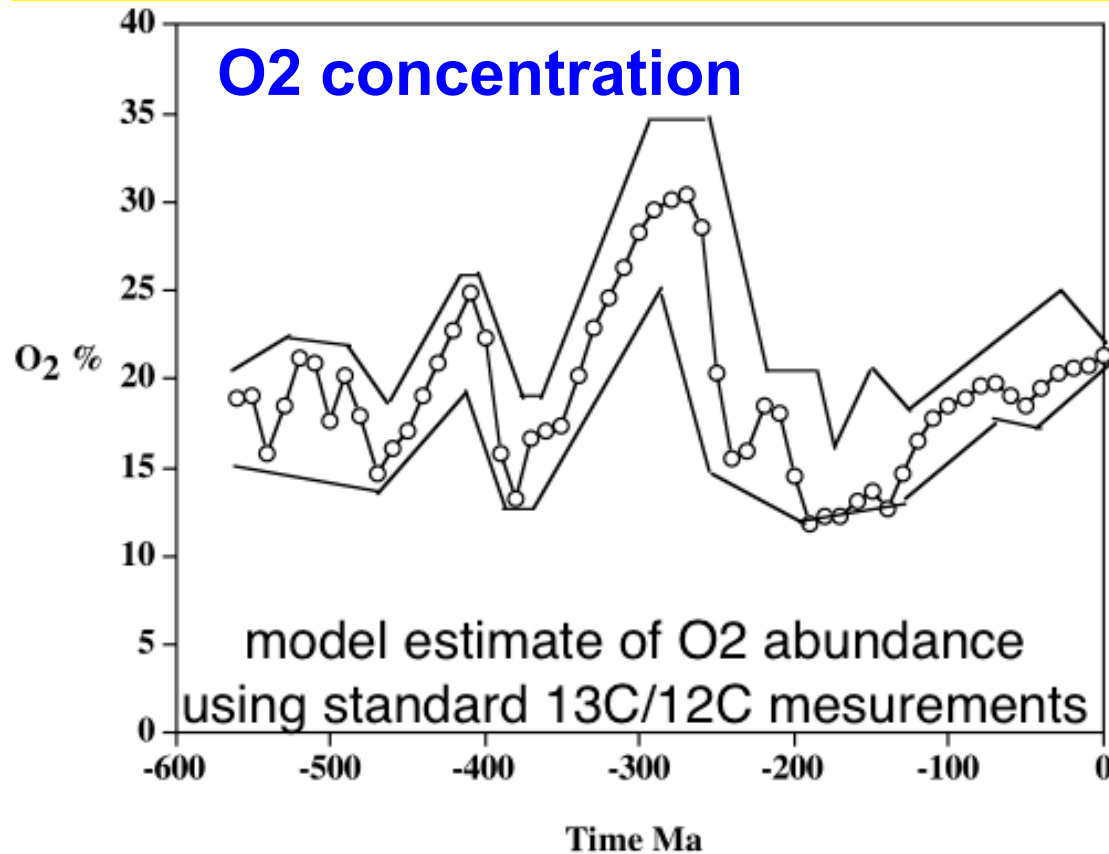


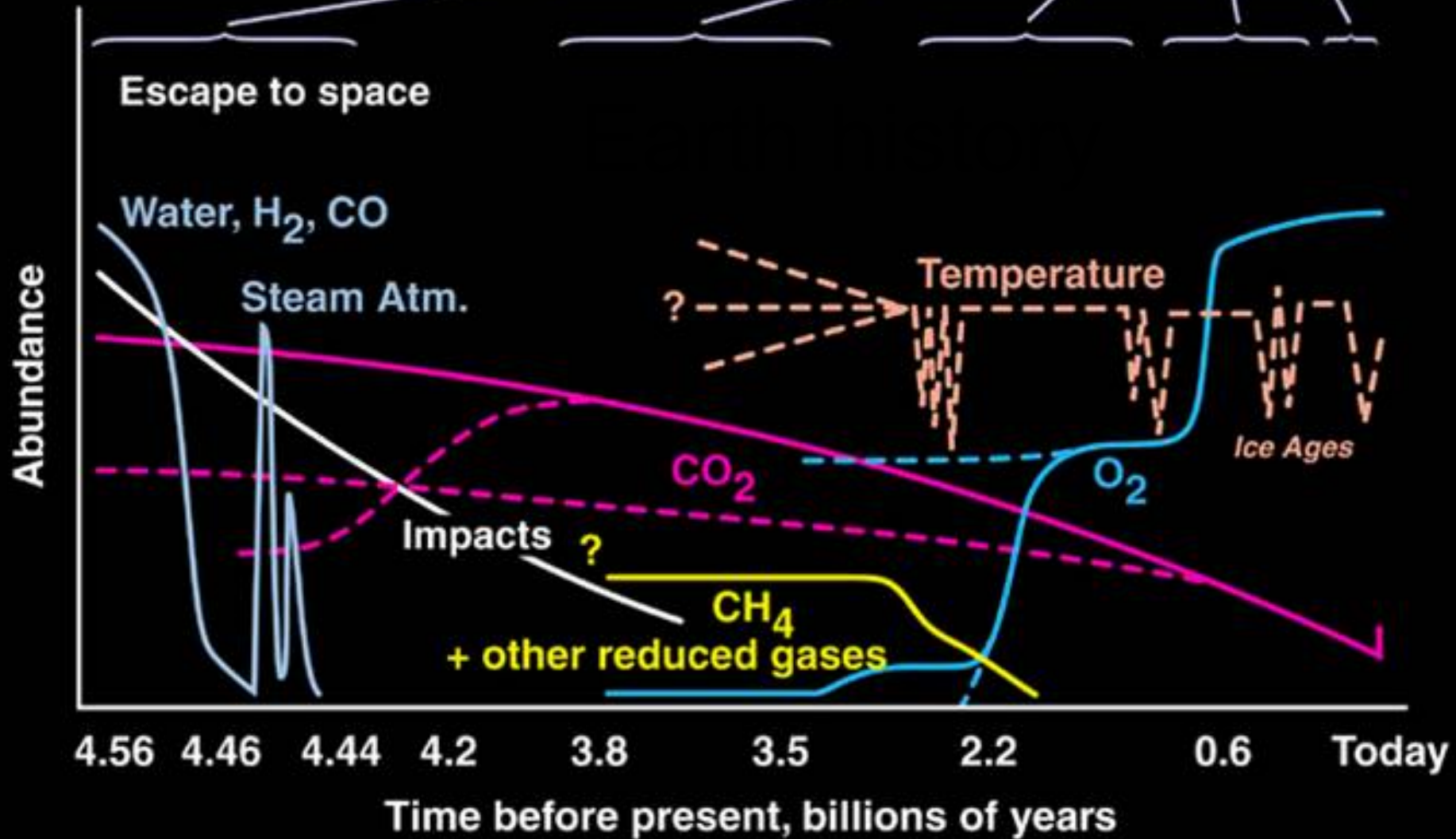
Fig. 20. Plot of O₂ vs time for the standard GEOCARBSULF model with

(*Berner, 2006*)

(*Halevy, 2017*)

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Earth's Atmosphere Through Time



Mission objective

To obtain the composition and flux of the atmospheric escape from the Earth and understand its effect on the atmospheric evolution.

- **The first-time exploration of the exosphere** to correctly model the thermal (inc. photochemical/hydrodynamic) escape.
- **Determination of the dominant non-thermal escape mechanisms** to evaluate relative importance of thermal and non-thermal escape.

To investigate fundamental parameters in space.

- **The first-time measurements of actual efficiencies** of the isotope fractionation and ionisation/neutralisation in the space environment, to estimate the past escape from a planet using the present atmospheric composition.

ESCAPE Scientific Objectives

How and at what rate is Earth slowly losing its atmosphere to space?

1. Build a quantitative and comprehensive picture for 500-2000 km altitudes

- Determine **exospheric** altitude density profiles and temperature profile **as a function of different drivers** such as **solar EUV, solar wind and geomagnetic conditions**.
- Establish **isotope ratios** for both **neutrals and ions** and compare them with those found at the Earth's surface and in other solar system objects.
- Determine **exospheric altitude profiles of ion/neutral ratios** and estimate ionisation / neutralisation efficiencies.
- Measure **temporal and spatial variations** of the density of **major exospheric species**.
- Correlate such **variability** with upper atmosphere parameters, and with different incident energies when **particle precipitation** is present.

2. Determine the dominant escape mechanisms, and their dependence on the drivers

- Estimate **thermal escape flux** for neutral and ion species **for different conditions**.
- Estimate the **prevailing escape mechanisms** and the relative importance of thermal or non-thermal escape **for different driver conditions**.
- Estimate the **response** of the ionisation / neutralisation efficiencies, isotope fractionation and the N/O ratio **to different drivers**.
- Estimate the **degree of recirculation of plasma** after it has left the ionosphere.

Escape Mechanisms of Heavy Ions/Neutrals

Type	Mechanism	Explanation
thermal	Jeans escape (H, He)	Thermal tail exceeds the escape velocity
non-thermal	Charge exchange	Heavy (trapped) ions collide with atoms
thermal	Momentum exchange	Light neutrals collide with heavy molecules
thermal	Hydrodynamic blow off (H, He)	Same as solar wind formation mechanism
chemical	(Photochemical heating)	Release of e.g. recombination energy of the excited state that accelerates the atom
combined	(Ion pickup)	Ions that are newly exposed to solar wind are removed by the solar wind ExB
non-thermal	(Atmospheric sputtering)	The energetic ions/atoms interact with the upper atmospheric molecules/atoms/ions
non-thermal	Large-scale momentum transfer & instabilities	Solar wind dynamic pressure and EM forces push the planetary plasma anti-sunward
non-thermal	Ion energisation by EM waves and $E_{//}$	EM disturbances / static E-fields energise ions, by e.g. the ion cyclotron resonance
non-thermal,	Plasmaspheric wind & plumes	Plasma instabilities near the plasmopause
non-thermal	Boundary shadowing	Drifting ions overshoot the magnetopause

Escape Mechanisms (continued)

Type	Mechanism	Where?	Δv for same T
thermal	Jeans escape	exobase ($M > E \geq V$)	$\exp()$
non-thermal	Charge exchange	above mirror altitude (E)	m^0
thermal	Momentum exchange	above exobase ($M > V$)	m^0
thermal	Hydrodynamic blow off	near exobase (ancient, all)	$m^{-0.5}$ to m^0
chemical	Photochemical heating	exosphere ($M > V \sim E$)	$m^{-0.5}$
combined	Ion pickup	outer exosphere (M, V)	m^0
non-thermal	Atmospheric sputtering	around and above exobase (M, $V > E$)	m^0
non-thermal	Large-scale momentum transfer & instabilities	magnetospheric boundary ($E > M \sim V$)	m^0
non-thermal	Ion energisation by EM waves and $E_{//}$	upper ionosphere & magnetosphere ($E > M > V$)	?
non-thermal,	Plasmaspheric wind, plasmaspheric plumes	plasmasphere ($E > M \sim V$)	m^0
non-thermal	Boundary shadowing	ring current (E)	m^0

neutrals

ions

What will be measured?

Basic altitude profiles in the exosphere & upper ionosphere (500–2000 km).

- **Temperature and densities** of major **neutrals and ions** (H, He, O, N, O₂, N₂, NO, and CO₂).
- **Isotope ratios** (¹⁷O/¹⁶O, ¹⁸O/¹⁶O, ¹⁵N/¹⁴N, D/H).
- Their **variability** in space and time (+ correlation with drivers)

Differential fluxes (energy-angle distributions) of the hot ions (N⁺, N₂⁺, O⁺, H⁺, and He⁺) in the lower magnetosphere and upper ionosphere > 800 km.

⇒ Correlation with external drivers: solar EUV flux, solar wind, and ionospheric/geomagnetic conditions.

What will be measured?

What to measure	Target range	Particle SI*	Other SI*
Density of major neutrals and cold ions, simultaneously	neutrals $1-10^6/\text{cc}$ ions $0.1-10^3/\text{cc}$	INMS, WCIMS	UVIS, SLP ASPOC
Density of minor isotopes (neutrals and ions)	neutrals $10^{-2}-10^3/\text{cc}$ ions $10^{-4}-10^1/\text{cc}$	INMS	SLP, ASPOC
Neutral temperature	500–1500 K	WCIMS	density +model
The energy distribution of major outflow ions	$10^{5-9} \text{ keV cm}^{-2} \text{ s}^{-1} \text{ str}^{-1} \text{ keV}^{-1}$	MIMS, NOIA	MAG, SLP
The flux of major returning energetic ions	$10^{6-9} \text{ keV cm}^{-2} \text{ s}^{-1} \text{ str}^{-1} \text{ keV}^{-1}$	EMS	MAG
The energy distribution of electron and photoelectron	$10^{7-11} \text{ keV cm}^{-2} \text{ s}^{-1} \text{ str}^{-1} \text{ keV}^{-1}$	ESMIE	MAG
Ionospheric auroral condition	10^{2-6} R		AMC
DC/AC field energy flux into to the ionosphere	$1-10^2 \text{ W/km}^2$		MAG, WAVES
Electromagnetic wave associated with ions	$0.1-10^3 \text{ Hz}$		MAG, WAVES
ENA flux for charge-exchanged keV ions	$10^{2-5} \text{ cm}^{-2} \text{ s}^{-1} \text{ str}^{-1}$	ENAI	

Why Earth?

Related to the past and future of the **habitable environment of the Earth**.

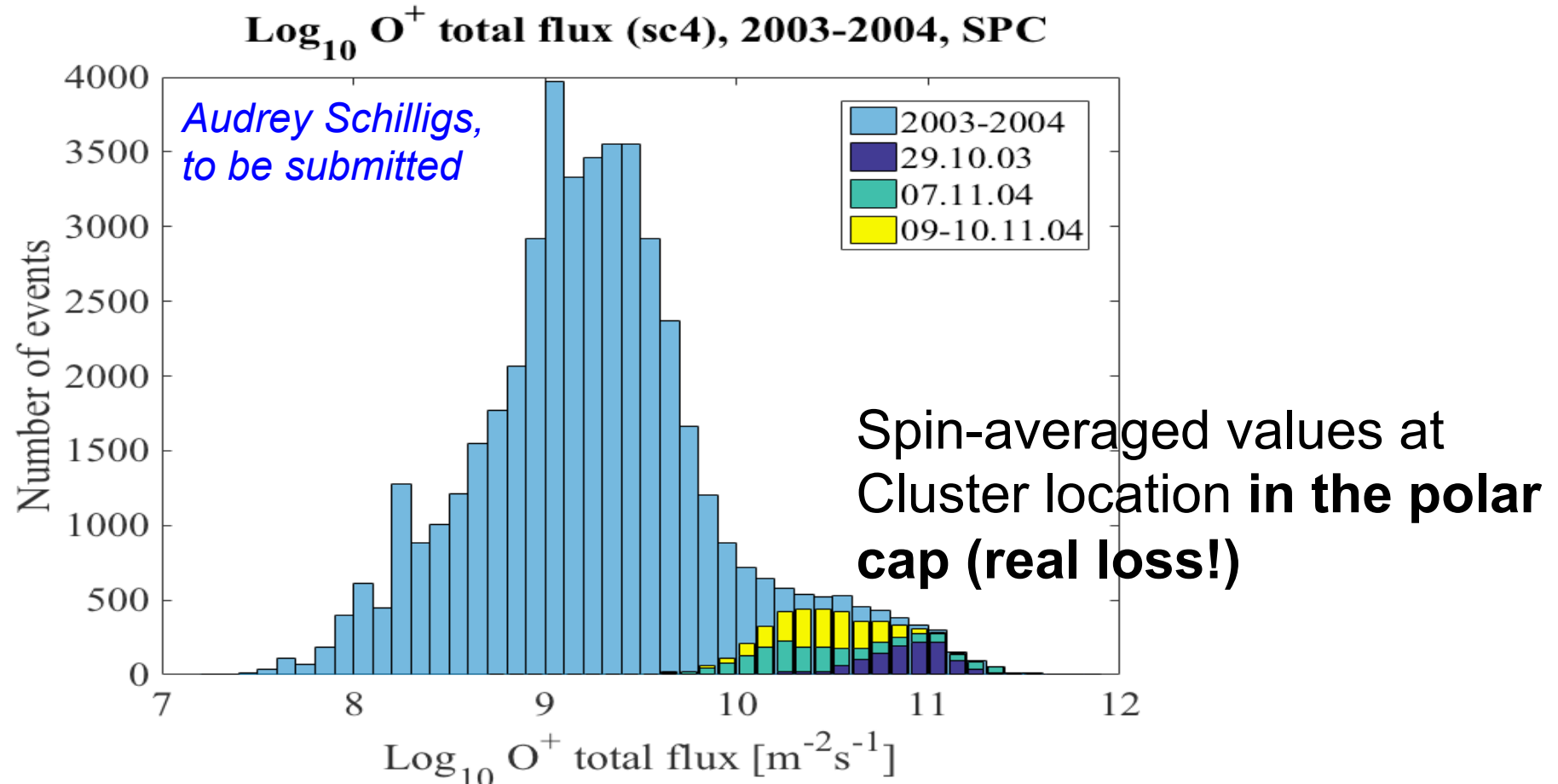
- **Rich database** (e.g., solar wind since 1960's) to allow probability assessment when **scaling back to the past**.

Can obtain the **best knowledge among all planets** thanks to many other measurements at and from the ground.

- Basically **common exospheric** structure with Mars and Venus.
- Sufficient **spacecraft resource** in the instrumentation and data transmission.
- Advantage of **ground-based** observations, e.g., EISCAT_3D.
- **Subsurface route** is best known.
- Some key area is less understood than Mars/Venus.

Why Earth? Amount matters

skip



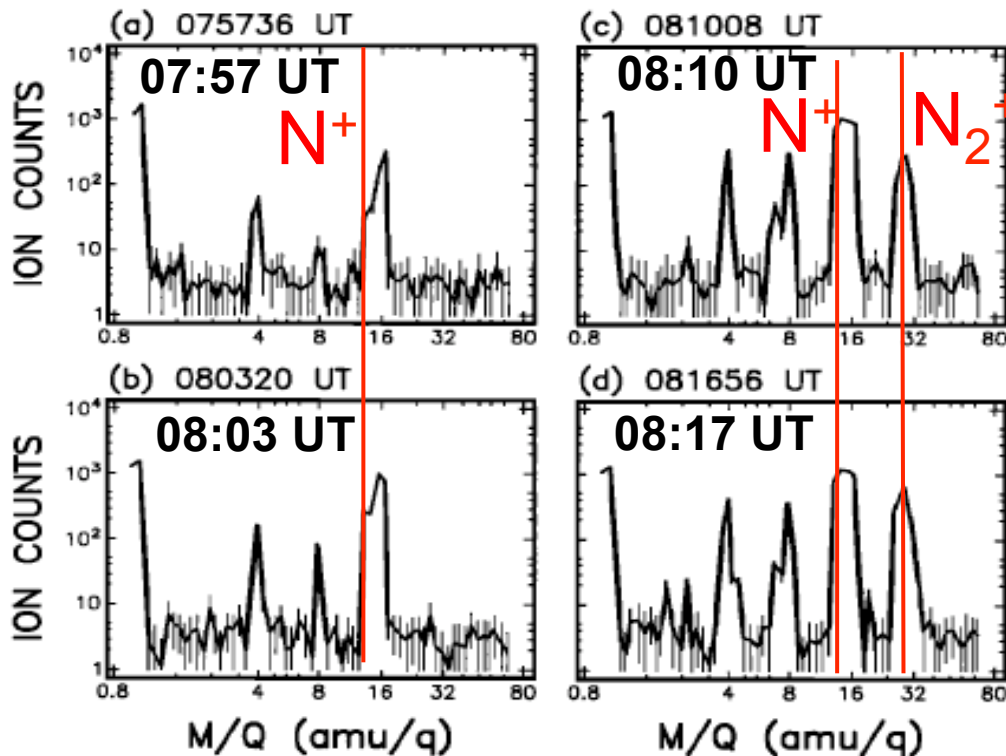
- Average O⁺ loss rate is about 10²⁵ ions/s.
- Increase by > 10 during storms.
- Increase by 10² during big event. ⇒ **10²⁷ ions/s during big events**

high N/O ratio in ancient time? skip

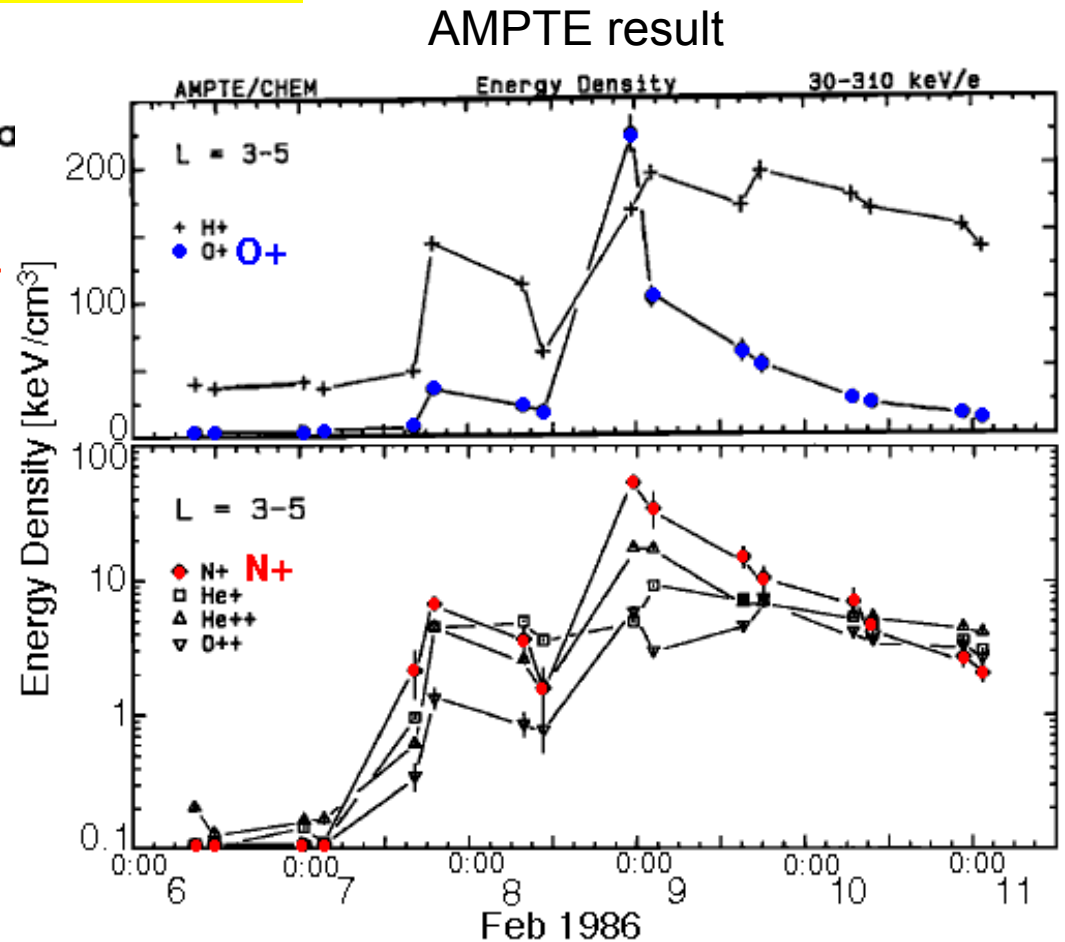
N/O ratio increases with activity

N^+/O^+ increase up to ~ 1

May 27, 1990 EXOS-D SMS NRC Canada



(Yau et al., 1993)



(Hamilton et al., 1988)

Small number of big events may account majority of N^+ escape

Why Earth? Amount matters skip

Biosphere responds to a few % change in the atmospheric O/N ratio (or water pH). \Rightarrow a 10% loss of nitrogen inventory in the biosphere ($\sim 5 \cdot 10^{18}$ kg $\sim 2 \cdot 10^{44}$) matters. $\Rightarrow 2 \cdot 10^{43}$ loss of nitrogen influences the evolution of life.

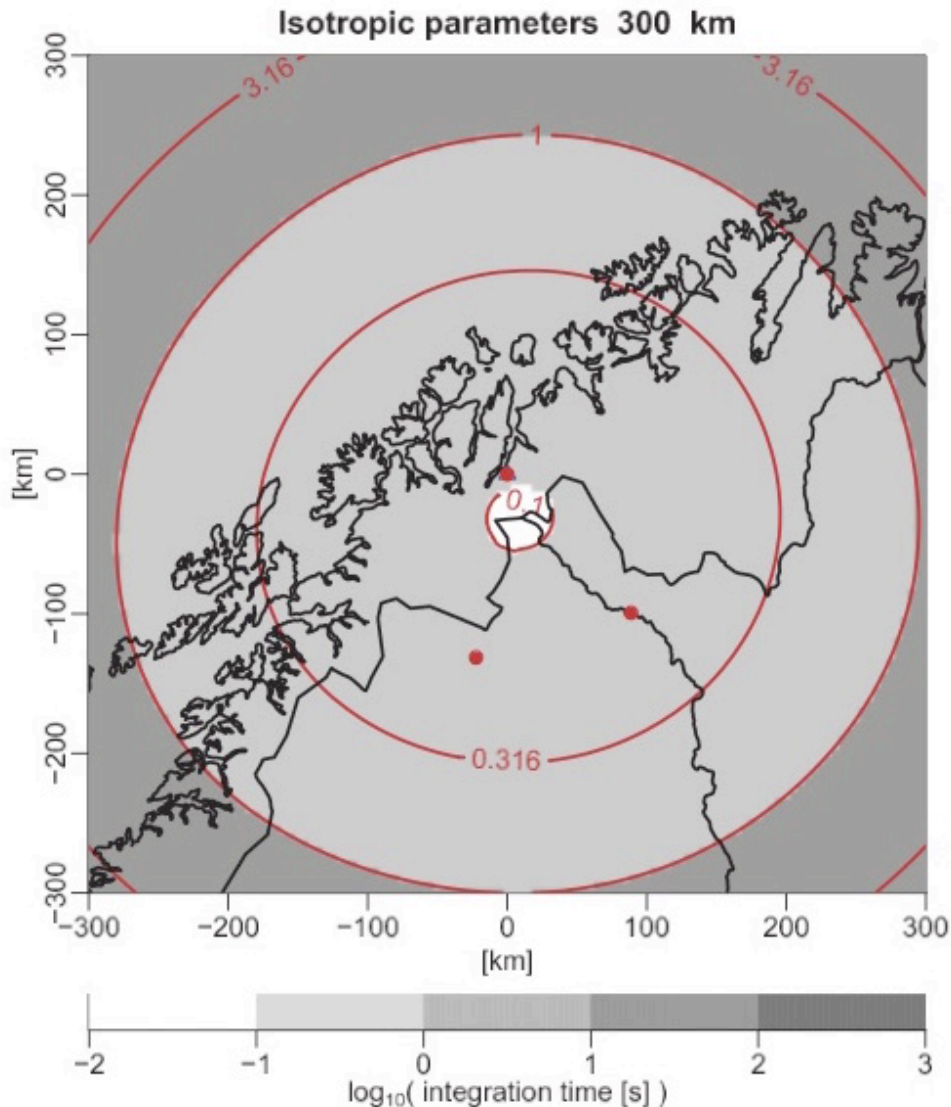
- **10^{27}s^{-1} x 600 million years ($2 \cdot 10^{16}$ sec) = $2 \cdot 10^{43}$.**
- **N/O ratio may reach to 1 during big events.**
- **Big events are good proxy for ancient time** (according to Sun-In-Time project) with **fast solar wind** and fast rotation (**strong solar dynamo**) for young G-type (Sun-time) stars.

Non-thermal escape (of N^+) alone could have influenced the biosphere

(in addition, subsurface circulation influences too).

Advantage of the Earth

Covering area of EISCAT_3D



High sensitivity in more than 500 km diameter (grey area) $\approx 15^\circ$ longitudinal range.



5 % of polar orbits traverses this region in average.

+

More conjugacy if we consider geomagnetic tracing.

+

Remote sensing can target this area from distance.

Why Exosphere?

Mandatory for thermal (inc. photochemical) escape modeling

(For thermal and some non-thermal escape)

Dynamic and structured

(Mars observation indicates unknown controlling factors other than EUV, e.g., atmospheric coupling)

Poor observational knowledge for the Earth

(No knowledge of > 800 km for nitrogen, > 1500 km for oxygen, and all altitude for isotope ratio)

Generate seeding cold ions through ionisation

(They contribute feeding ions of non-thermal escape)

Determine charge-exchange escape of trapped ions

(Including magnetopause shadow, majority of return flow from the tail will finally escape, and important input in estimating ENA flux)

Exosphere = thermal/photochemical

Mandatory for thermal (inc. photochemical) escape modeling

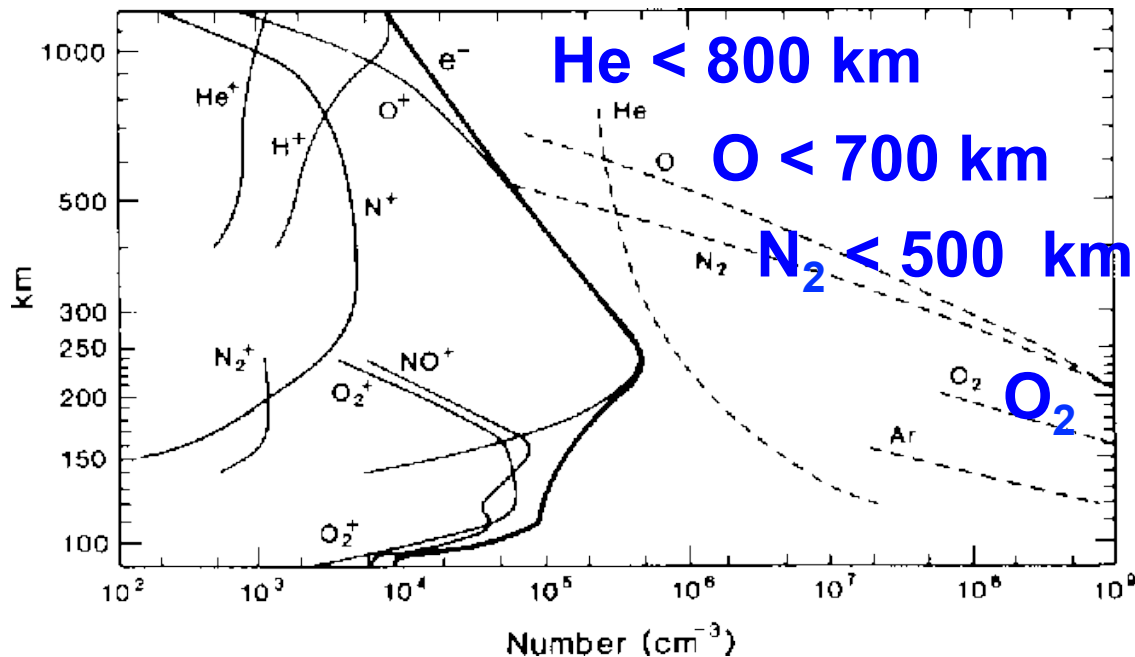
height	surface	500 km			2000 km		
		velocity	O (eV)	N (eV)	H ₂ (eV)	O (eV)	N (eV)
Earth	11 km/s	9.7	8.	1.2	8.0	7.0	1.0
Venus	10 km/s	8.2	7.2	1.0	6.7	5.8	0.8
Mar	5.0 km/s	1.8	1.6	0.23	1.3	1.1	0.16
Moon	2.4 km/s	0.5	0.4	0.06	0.31	0.27	0.04

before		after	extra energy
O ₂ ⁺ + e ⁻	-->	2O	3–7 eV
N ₂ ⁺ + e ⁻	-->	2N	3–6 eV
N + O ₂	-->	NO + O	2–4 eV

Why Exosphere? poor knowledge

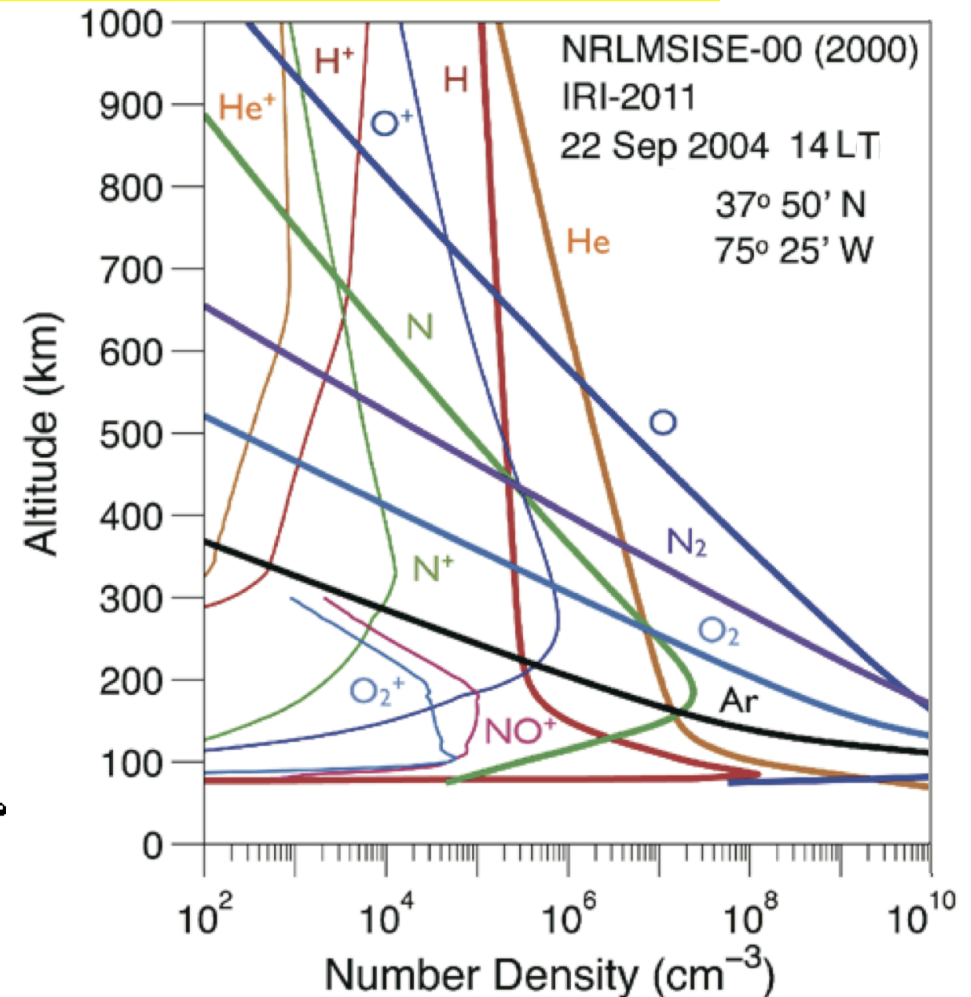
No systematic in-situ observations > 800 km

(except H)



In-situ measurements are really old, and we need to update.

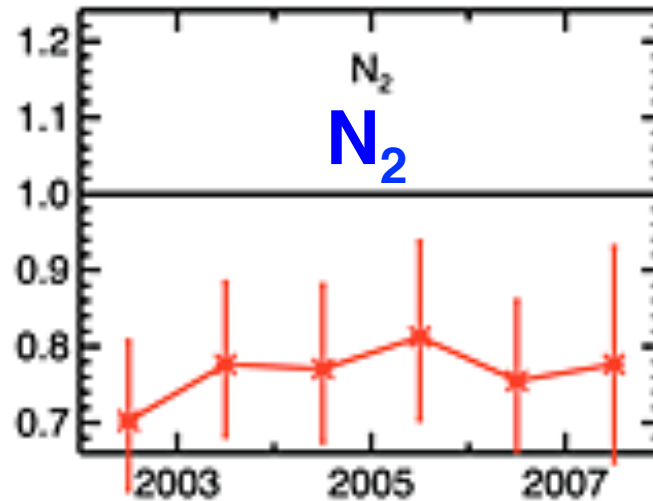
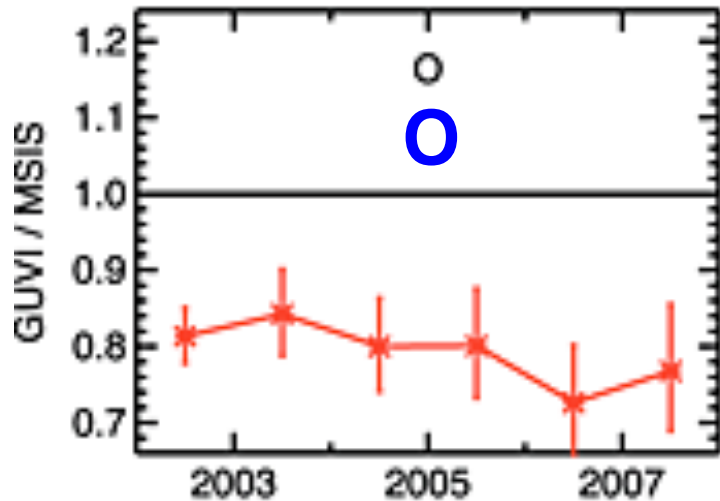
(Johnson, 1969)



MSIS-2000 model (smooth > 200 km)

(Pfaff, 2012)

Why Exosphere? poor knowledge

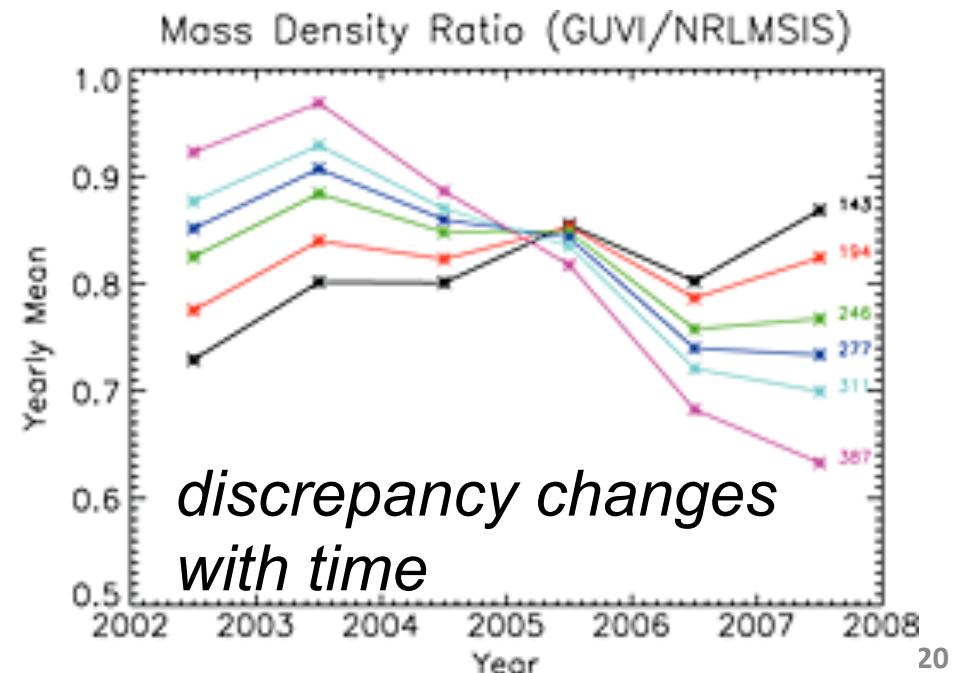


(TIMED) GUVI

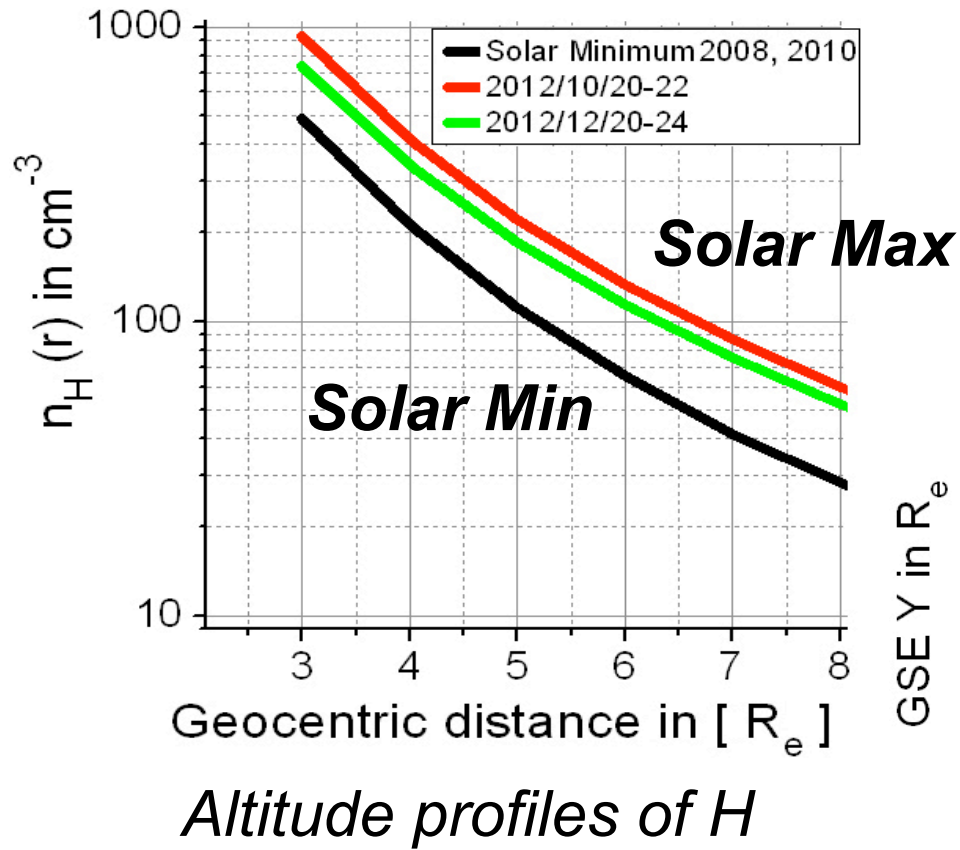
GUVI obs. vs. MSIS model

**Even the lower altitude
(thermosphere) is not well known**

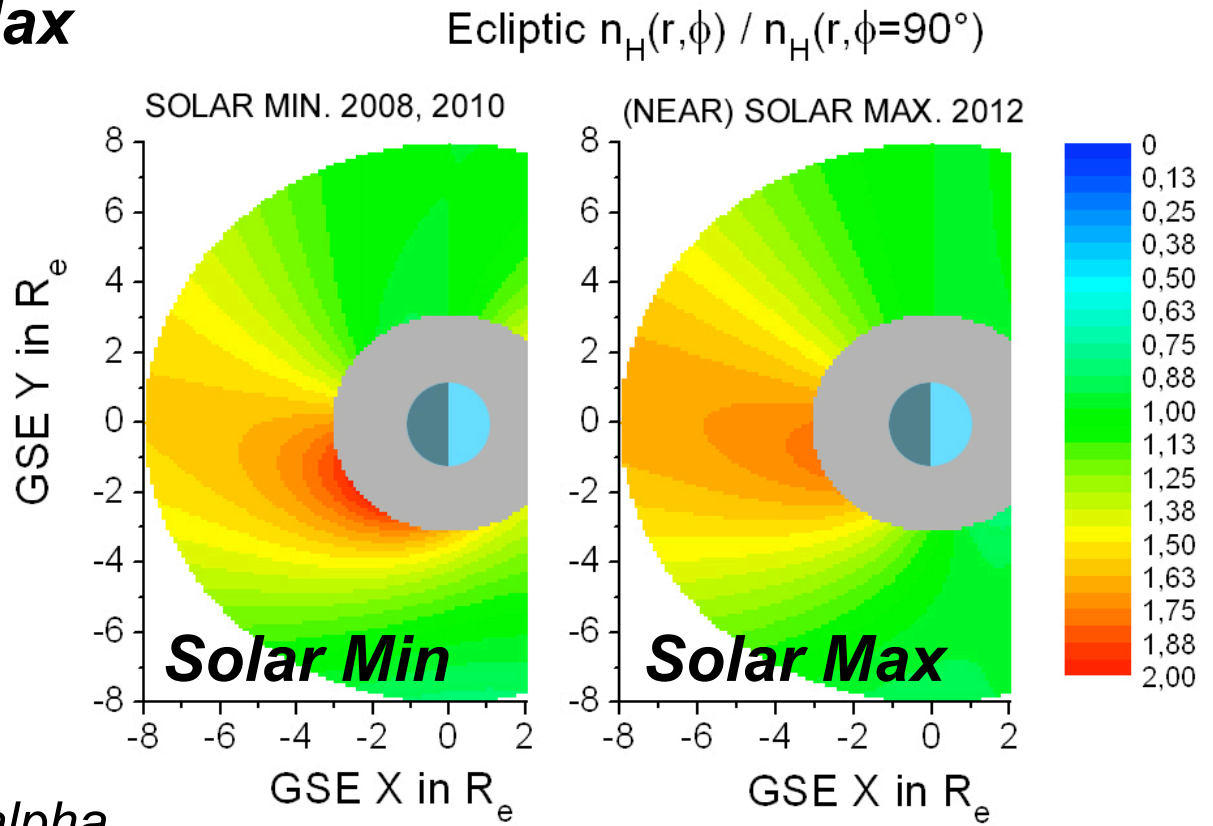
*Left: GUVI / NRLMSIS ratio at 194 km altitude for geomagnetically quiet condition.
Right: GUVI / NRLMSIS ratio of mass at different altitudes from 143 km to 387 km (Meier et al., 2015).*



Exosphere: structured & dynamic



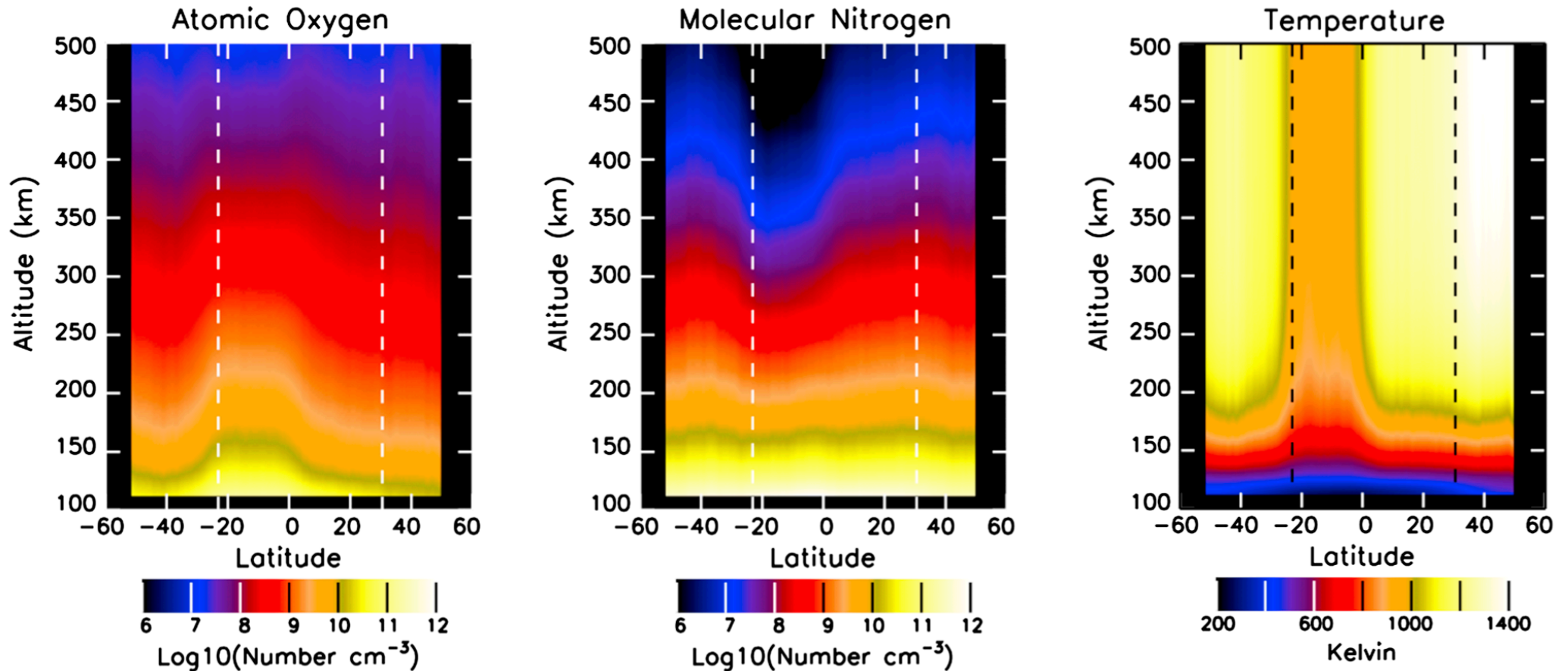
(TWINS) Lyman-alpha



Model-fitted H density from Lyman-alpha line-of-sight stereo observations by the two TWINS spacecraft (Zoennchen et al., 2015)

Nightside distribution of H

Exosphere: structured & dynamic



GUVI: 2004 DOY209 ORBIT14273 all SZA

Model-fitted GUVI observation for 27 July 2004. (Meier et al., 2015) during magnetic storm ($F_{10.7}=128$, $A_p = 186$). Temperature increased summer auroral region

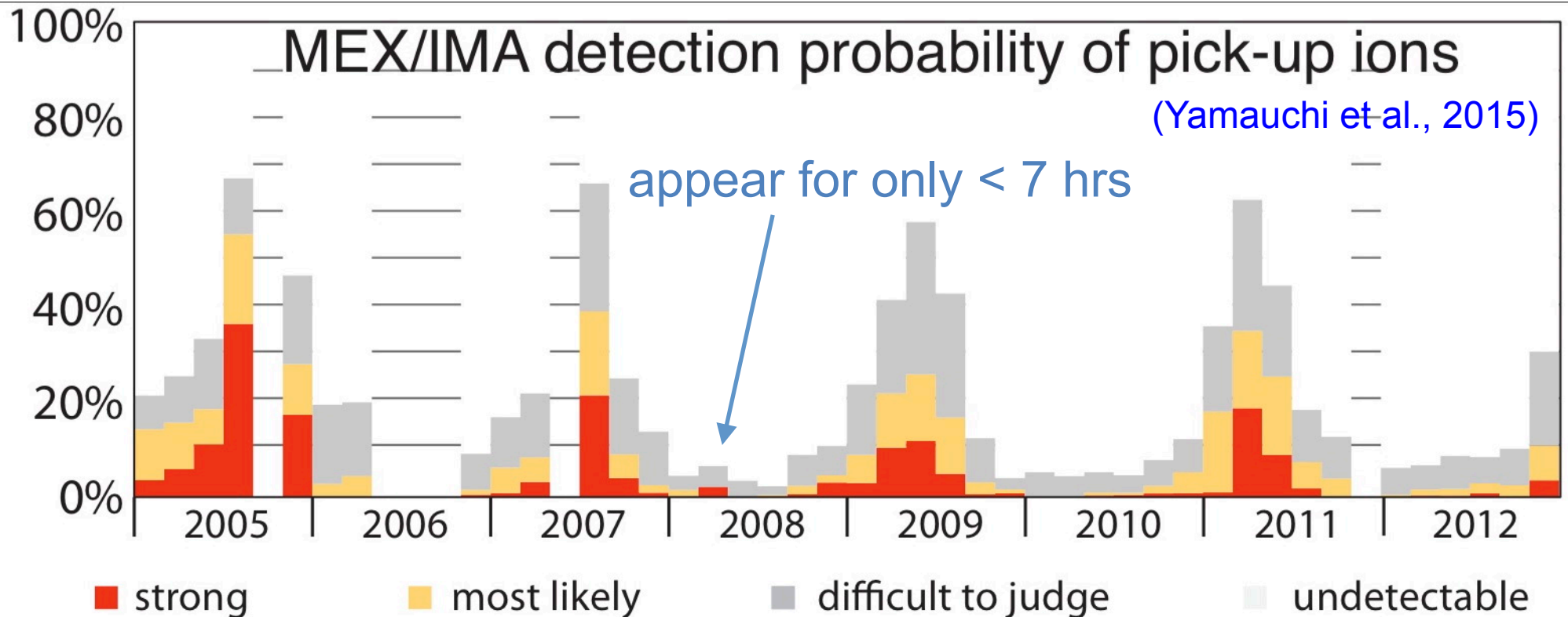
No dedicated Exospheric missions

Mission	UV instrument	Particle instrument
IMAGE	Lyman-alpha	ENA
TWINS	Lyman-alpha	ENA
GOES	Lyman-alpha	hot ions
TIMED	multi-wavelength (thermosphere)	
Hisaki	multi-wavelength (planetary)	
Polar		hot ions
Cluster		hot ions
Akebono		cold ions with N/O separation, hot ions
e-POP		cold ions with N/O separation

Why exosphere?

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(Mars Express) newly-formed cold ions



- >10 ion flux change by 20% change in Sun-Mars distance
- >3 ion flux change within 7 hours

Why isotope ratio?

Used as indicator of (thermal) escape from planet, but only qualitatively

(isotope ratios are different between different escape processes)

Gives important information on hydrostatic thickness

(the dependence is also exponential to the mass x thickness)

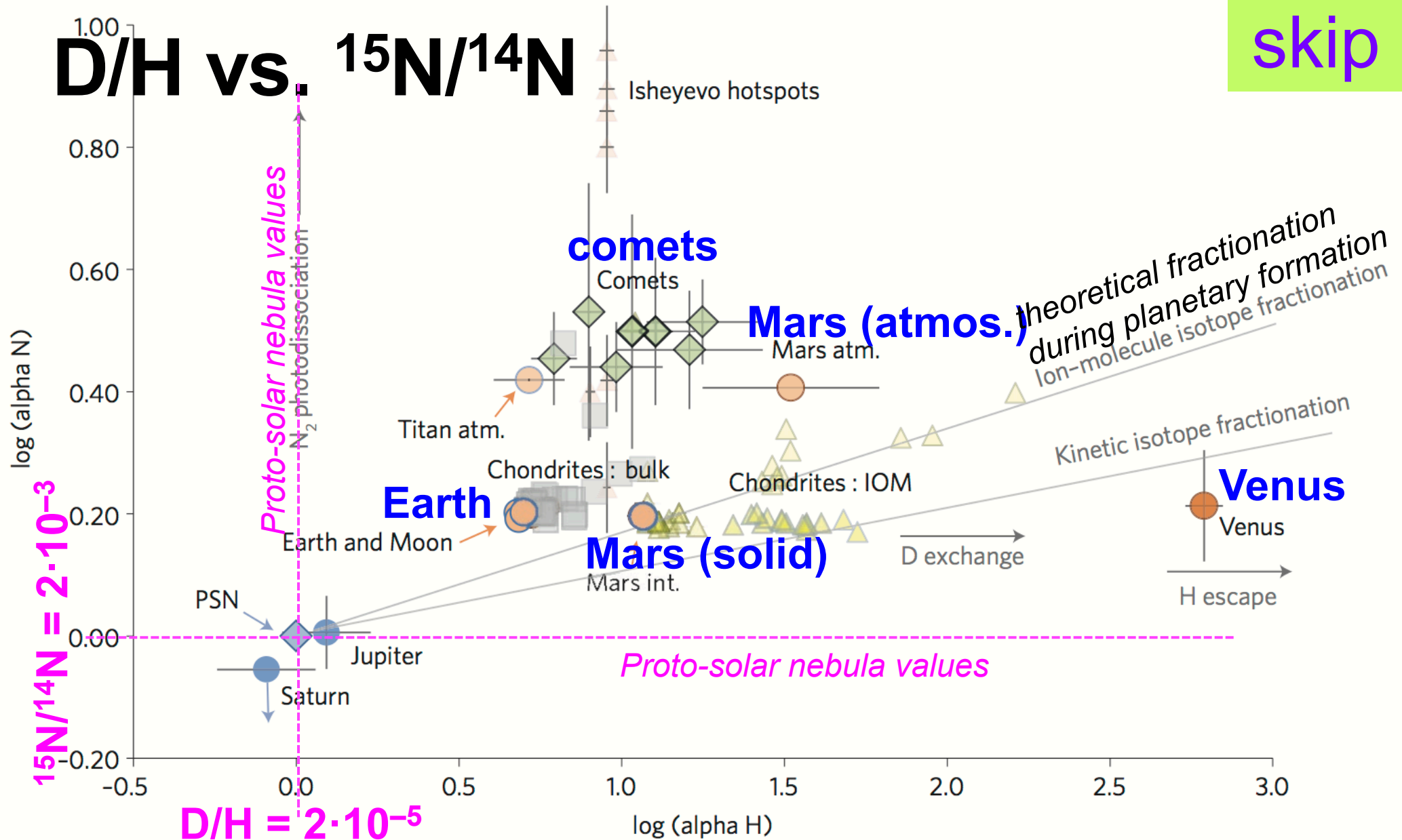
⇒ **We expect both temporal and horizontal variation**

(MAVEN detect it)

Poor knowledge for the Earth's upper atmosphere

(no systematic observation in the magnetosphere/ionosphere for seeding population, and the mass-filtering depends on ionization altitude)

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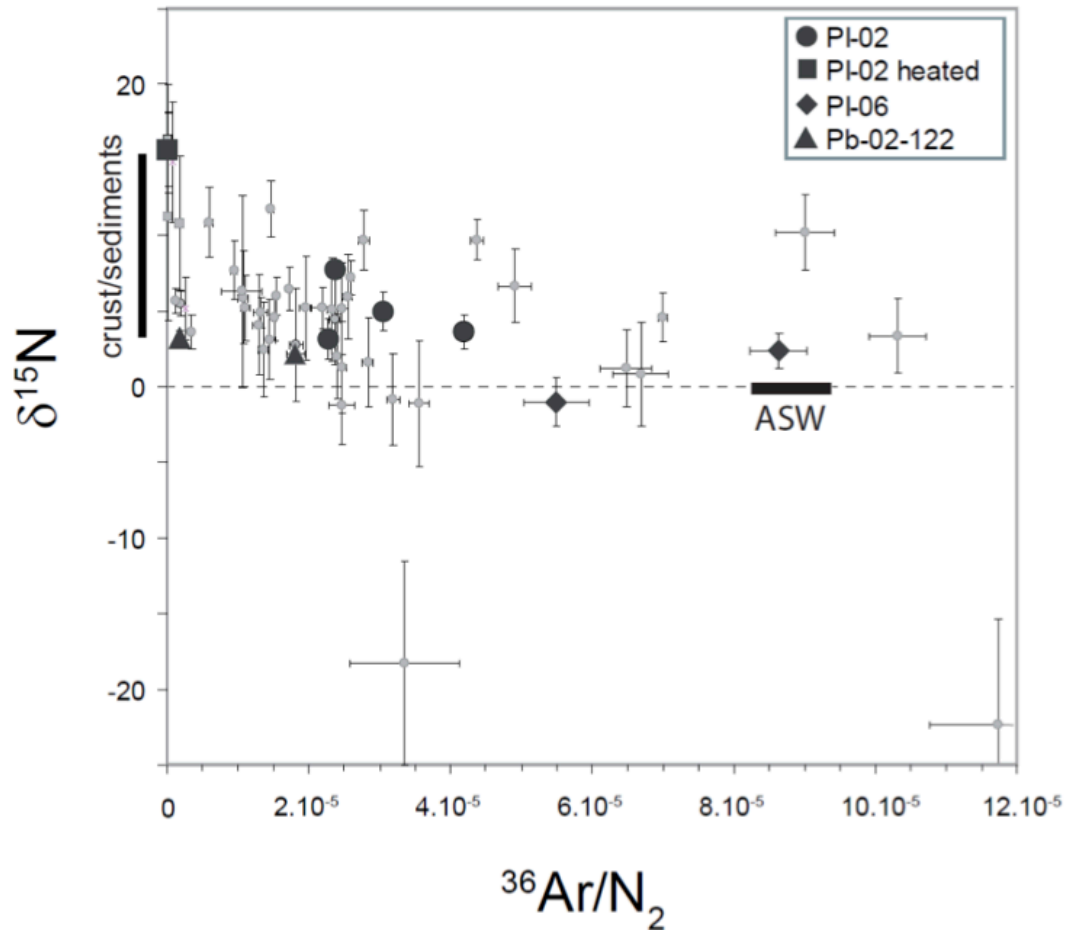


Same $^{15}\text{N}/^{14}\text{N} \Rightarrow$ same initial composition

(Füri and Marty, 2015).

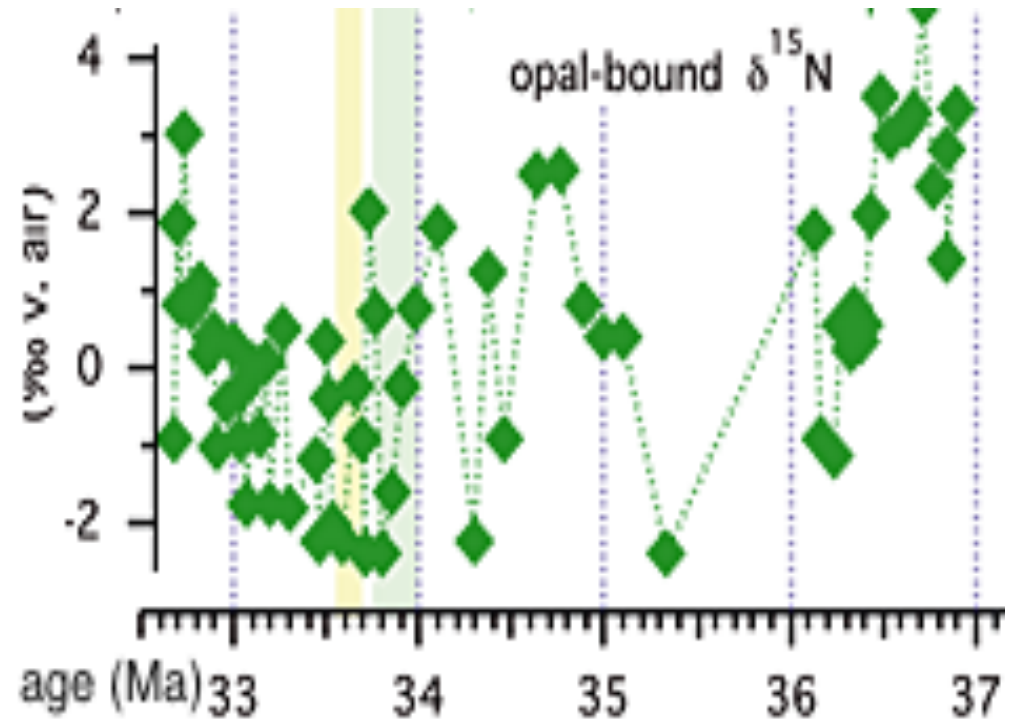
Alpha H and alpha N are deviations of D/H and $^{15}\text{N}/^{14}\text{N}$ from protosolar nebula values.

constant $^{15}\text{N}/^{14}\text{N}$ ratio



(Marty et al., 2013)

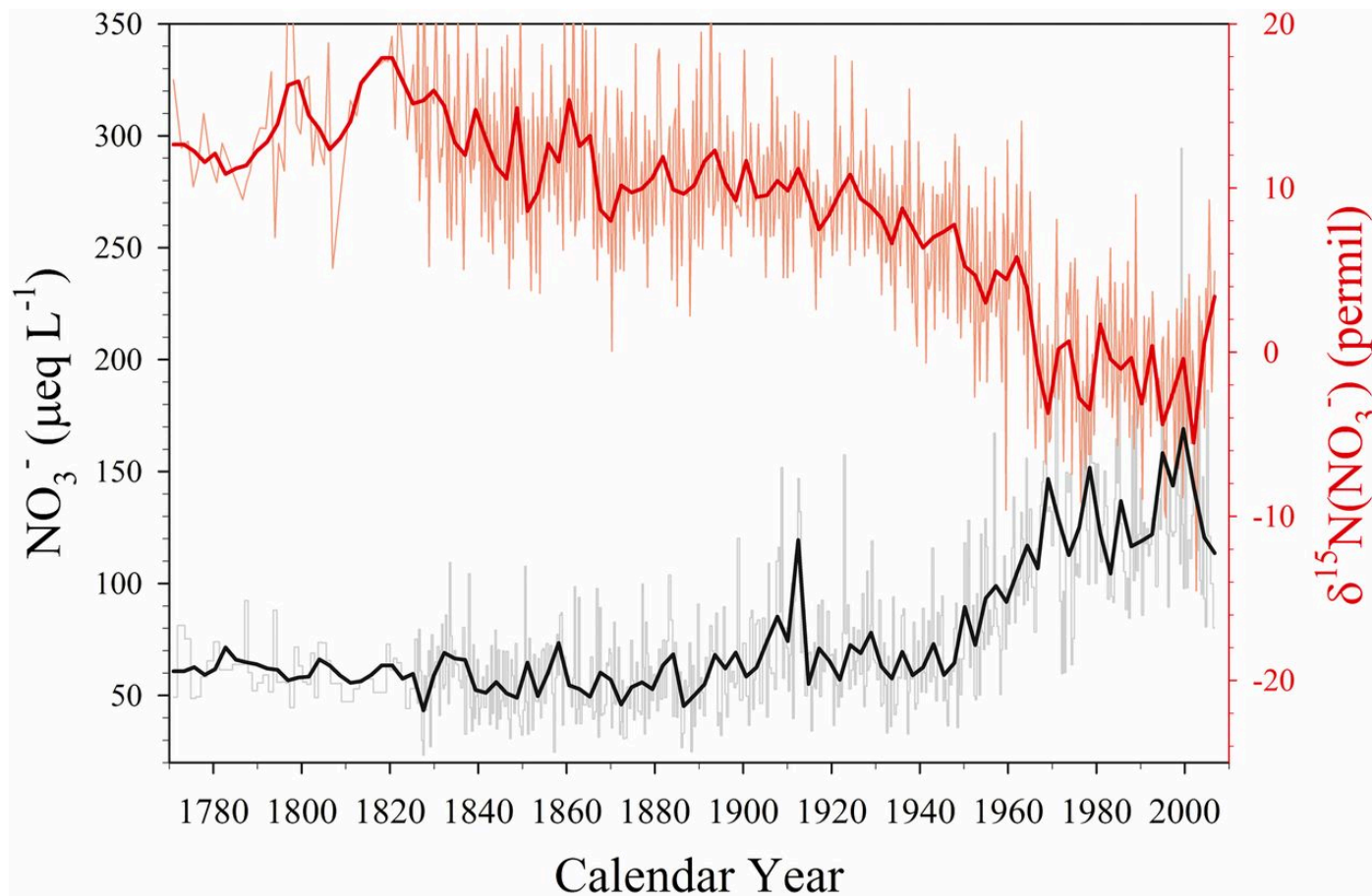
changing $^{15}\text{N}/^{14}\text{N}$ ratio



(Robinson et al., 2015)

$^{15}\text{N}/^{14}\text{N}$ ratio ~ acid level

Ice core observations of NO_3^- past 200 years



(Geng et al., 2014)

Why isotope ratio?

Hydrostatic pressure scale heights h is inversely proportional to the mass of atoms/molecules.

⇒ H, D, ^{16}O and ^{18}O , $h_{\text{H}} = 2 \cdot h_{\text{D}} = 16 \cdot h_{^{16}\text{O}} = 18 \cdot h_{^{18}\text{O}}$
(where $h_{\text{H}} \sim 300$ km for the Earth's ionosphere)

⇒ when T is constant,

$$n_{\text{D}}/n_{\text{H}} \sim \exp(-z/h_{\text{H}}) \quad \& \quad n_{^{18}\text{O}}/n_{^{16}\text{O}} \sim \exp(-2 \cdot z/h_{\text{H}})$$

⇒ **Scale height of isotope ratio \sim scale height of H.**

The thickness of the hydrostatic layer (or degree of vertical convection) is one of the leading factor that **determines the isotope ratio** (cf. Table 1).

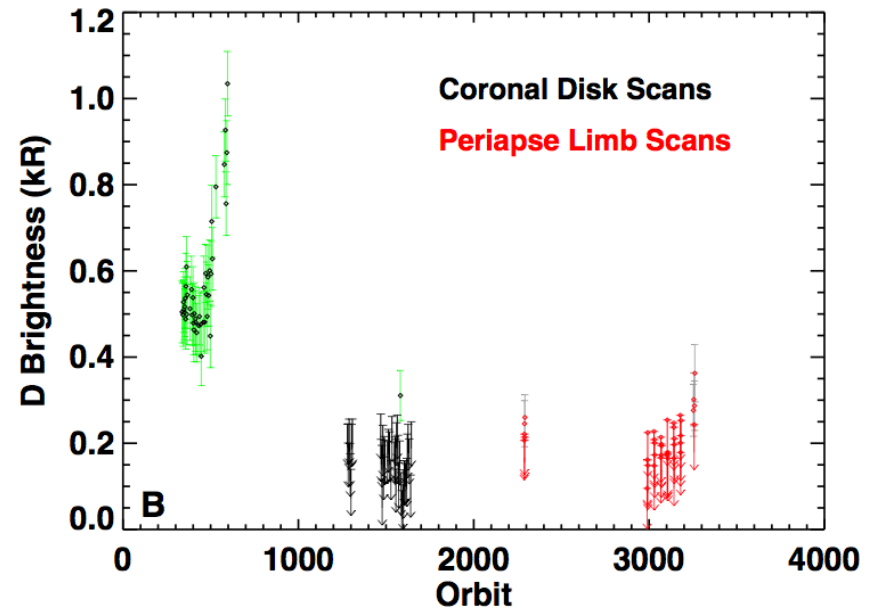
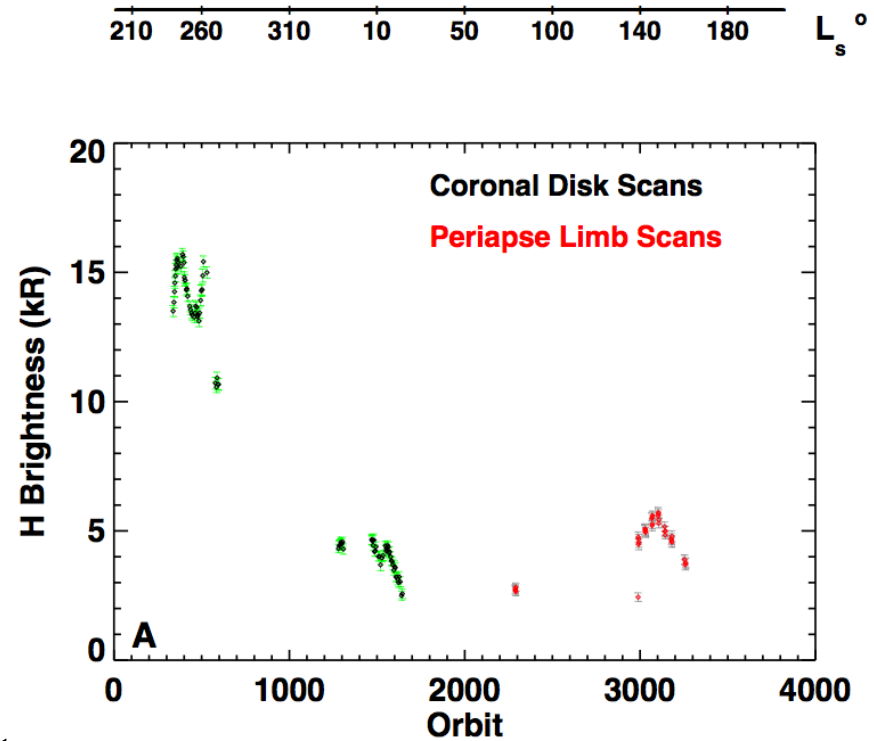
⇒ **Isotope ratio can be dynamic and structured.**

Isotope ratio is dynamic

(MAVEN) UVI

The variations of H and D are different, indicating changes in the D/H ratio in time.

MAVEN observation of H and D in the Martian exosphere (Courtesy by John Clark; Clark et al., 2016, submitted to JGR).



Non-thermal escape?

Evaluate the composition change by escape

⇒ **Relative contribution of thermal & non-thermal escapes**
(They have different mass-filtering effect)

At moment, we do not know if they are related (they might even anti-correlate to each other, but who knows?)

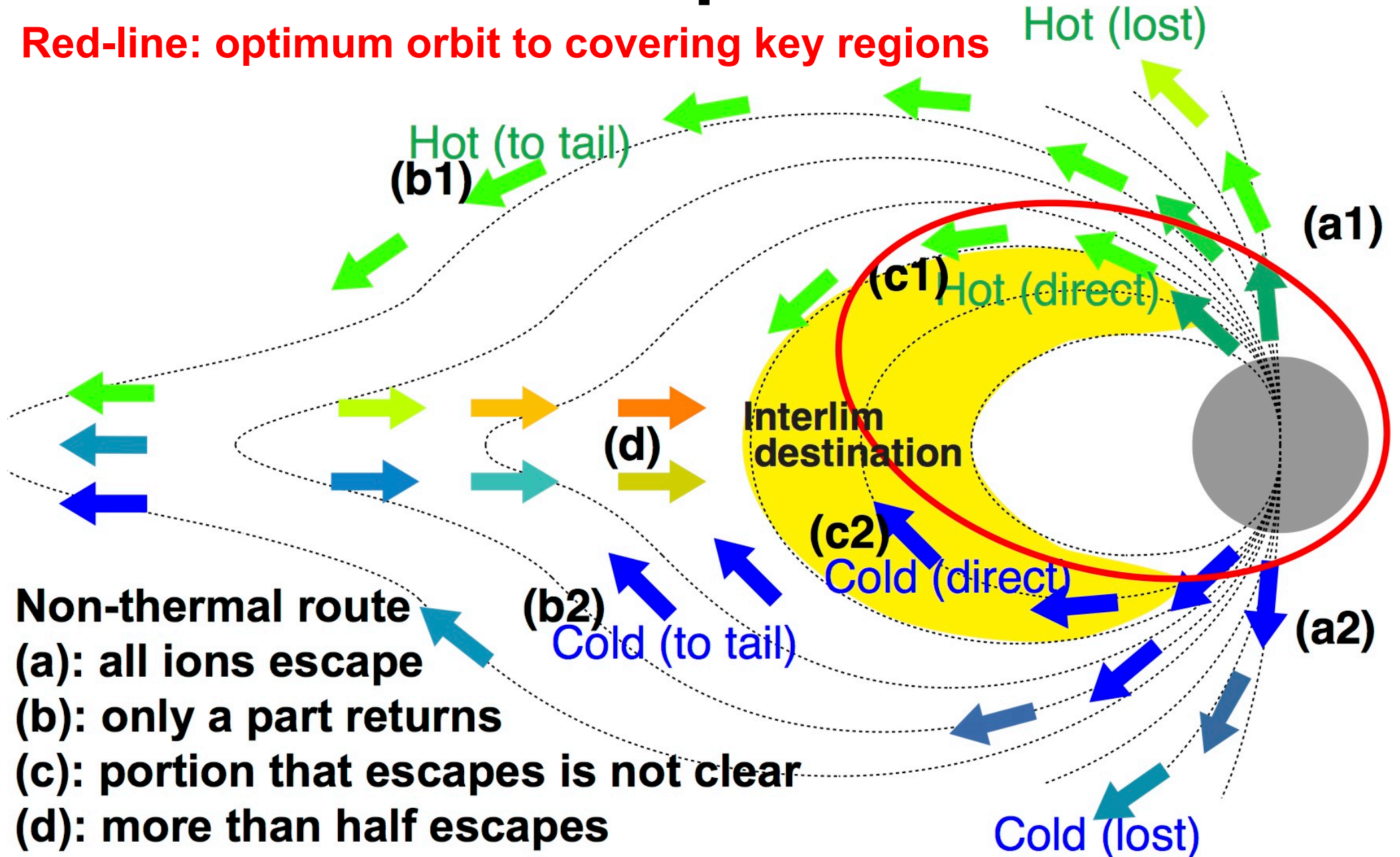
⇒ **Need simultaneous observation**

No systematic observation at the Earth

(all observations have been done instrument-level but not mission level)

Where is the optimum orbit?

Red-line: optimum orbit to covering key regions



Fate of ions in each route

- (a) Polar outflow to **solar wind**: **all ions escape** ($O^+ \sim 10^{25} \text{ s}^{-1}$)
cf. Prognoz-7, but not enough statistics → **Cluster answered**.
(Hawkeye without composition)
- (b) Polar outflow to **magnetotail**: **some escapes** ($O^+ \sim 10^{24-25} \text{ s}^{-1}$)
cf. Geotail, but too many assumptions → **Cluster corrected**.
- (c) Auroral/subauroral outflow to **inner magnetosphere & plasmasphere**: portion that escapes is **not clear for O^+**
cf. Geosynchronous, DE-1, etc., but not enough coverage
→ **Cluster found it could be $H^+ \geq 10^{27} \text{ s}^{-1}$ in average**.
- (d) Return from magnetotail: **>50% escapes** ($O^+ \sim 10^{24-25} \text{ s}^{-1}$)
cf. simulations + many past missions → **Cluster confirmed**.

Summary of amount

(a) polar outflow

Satellite	R (Re)	Energy Range keV	Outflow Rate (10^{25} s^{-1})		References
			H ⁺	O ⁺	
Cluster (CIS)	> 8	0.028-40		escape ~ 2	Nilsson et al. 2012 Slapak et al. 2017
Cluster (EFW)	> 10	supersonic	tail ~ 10		Engwall et al. 2009
Polar (Apogee)	8.5	0.015–33	1	1.5	Lennartsson et al. 2004
DE-1	4	0.010–17	1	0.7	Yau et al. 1986 Collin et al. 1988
Polar (Perigee)	2-2.3	0.015–33	0.3	0.3	Peterson et al. 2001
Akebono	2.3	< 0.07	2	0.2	Cully et al. 2003a
FAST	1.4	0.003–12	-	0.2	Andersson et al. 2005

(b) magnetotail ($\times 10^{25} \text{ s}^{-1}$): earthward O⁺ ~ 0.6 / tailward ~ 0.5 / net return ~ 0.1

(c) inner magnetosphere / Cluster

Instrument	phenomenon	method	escape rate (10^{25} s^{-1})	References
WHISPER	Plume	ω_{ce}	peak H ⁺ > 100	Darrouzet et al. 2009
CIS (cold)	Wind	direct	H ⁺ ~ 50	Dandouras et al. 2013
CIS (hot)	Heating	direct	H ⁺ < 0.01	Yamauchi et al. 2012

(d) returned flow: about half are lost < $0.1 \cdot 10^{25} \text{ s}^{-1}$

key region to measure

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Exosphere

Ionosphere

(Ionospheric condition should influence non-thermal escape, and need to monitor even lower part simultaneously.)

Plasmasphere

(Plasmasphere could be partly filled by the exosphere directly)

Magnetosphere

(energy coming from the magnetosphere must be major external force as well as solar UV, and some outflow returns.)

Solar wind

(Essential external condition Continuously observed)

Particular interest on N^+ and N_2^+ skip

Good mass filtering test for non-thermal escape

($M/dM > 700$ works only for cold ions, and need other method to extract mass-filtering by the non-thermal mechanism)

Behavior is different from Oxygen with similar mass

($N_2 \rightarrow N_2^+ \rightarrow N^+$ whereas $O_2 \rightarrow O \rightarrow O^+$, and completely different solar/geomagnetic dependence)

dissociation energy for molecules:

N_2 : 945 kJ/mole (9.79 eV) ↪

O_2 : 497 kJ/mole (5.15 eV) ↪

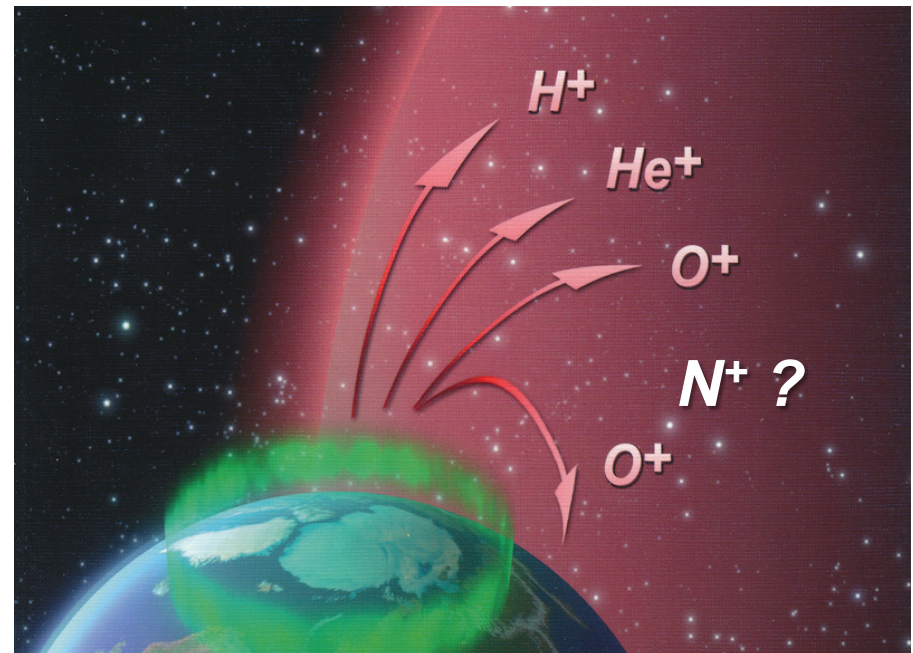
H_2 : 436 kJ/mole (4.52 eV)

Ionisation energy for atmos:

N: 1402 kJ/mole (14.53 eV)

O: 1314 kJ/mole (13.62 eV)

H: 1312 kJ/mole (13.60 eV)



Summary of Scientific Objective

#1: What is the quantitative state of the atmosphere at altitudes of 500-2000 km?

- Exospheric temperature and density altitude profiles as a function of different drivers (solar EUV, solar wind and geomagnetic conditions).
- Establish isotope ratios for both neutrals and ions and compare them with those found at the Earth's surface and in other solar system objects.
- Exospheric altitude profile of ion/neutral ratios and estimate ionisation/neutralisation efficiencies
- Temporal and spatial variations of the density of major exospheric species
- Correlation of such variability with upper atmosphere parameters, and with different incident energies when particle precipitation is present

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Summary of Scientific Objective

#2: What are the dominant escape mechanisms, and their dependence on drivers?

- Estimate thermal escape flux for neutral and ion species for different conditions.
- Estimate the prevailing escape mechanisms and the relative importance of thermal or non-thermal escape for different driver conditions.
- Estimate the response of the ionisation and neutralisation efficiencies, isotope fractionation and the N/O ratio to different drivers.
- Estimate the degree of recirculation of plasma after it has left the ionosphere.

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Summary of Scientific Objective

#3: How are fundamental physics/chemistry processes affected by the space environment?

- How are isotopes fractionated in the space environment?
- What are the ionisation/neutralising efficiencies in the space environment?

skip

Observation strategy

#1: Density and temperature distribution of Exosphere

- Combine in-situ and line-of-sight integrated observation (particle instrument + UV imaging spectrometer)
- Measure ions and neutrals simultaneously

#2: Both thermal and non-thermal

- Elliptic orbit with high-inclination
- Return flow is also measured

#3: External/internal drivers' condition as much as possible

- Monitor ionospheric & magnetospheric conditions
- Take advantage of EISCAT_3D

Spin-off

skip

Correction of exospheric contamination/abruption in optical (astrophysical) observation

(cf. Hisaki background from 1000 km altitude is very high)

Ionospheric physics

(together with e.g., EISCAT_3D)

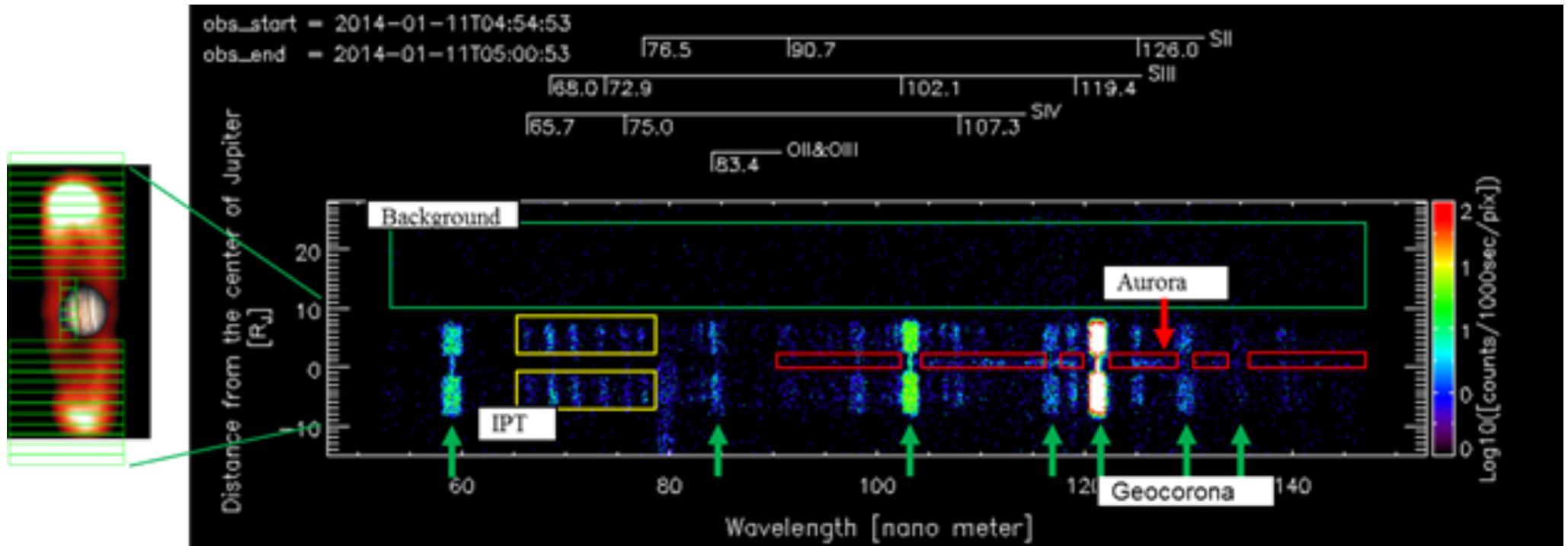
Exospheric effect on Ionosphere-Plasmasphere coupling

(through ionisation as well as neutralisation)

Magnetosphere-ionosphere coupling

(covering both region + N^+/O^+ difference as tracer of different start time)

Exospheric contamination (Hisaki)



An example (a 6 min cycle) of EUV (50-150 nm) observation of Jupiter by Hisaki (Yoshikawa et al., GRL, in press).

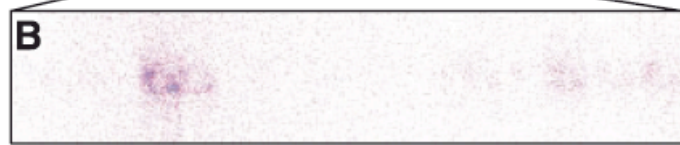
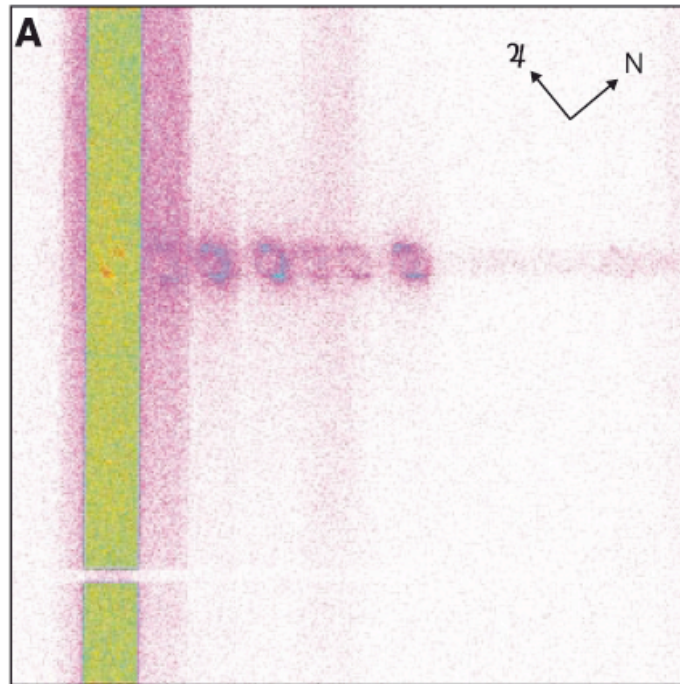
Green = emission from exosphere

Yellow = emission from Io plasma torus

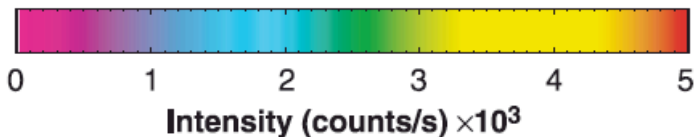
Red = Jovian aurora

skip

HI: 1216
OI: 1304, 1356, 1667
SI: 1251, 1299, 1389, 1479, 1429
SII: 1204, 1256
SIII: 1194, 1712
SIV: 1406, 1425



OI: 1356, 1359
SI: 1386, 1389, 1396, 1382, 1388, 1393

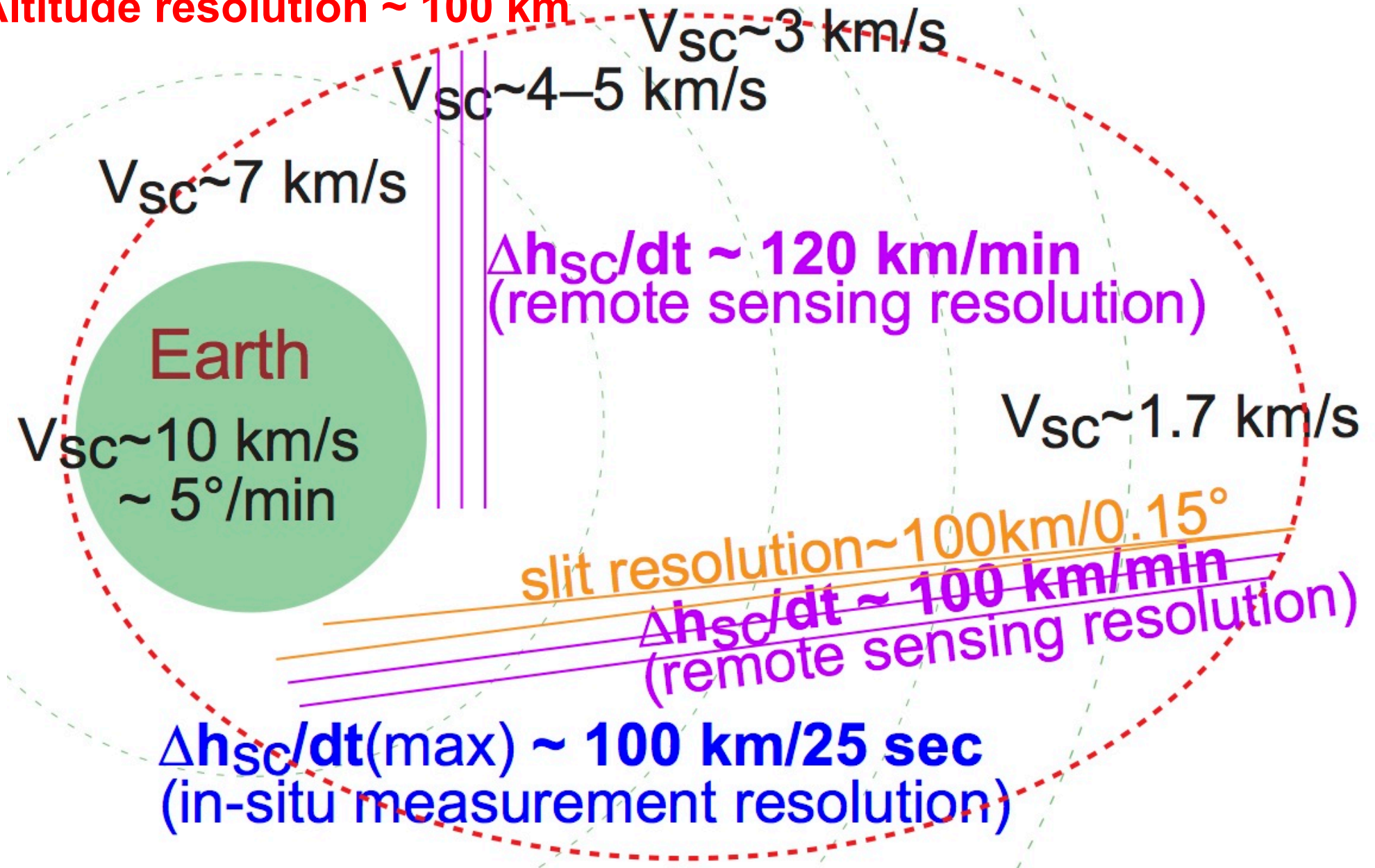


exospheric contamination/ absorption

Example of geocoronal noise in an observation taken with the Space Telescope Imaging Spectrograph onboard the Hubble Space Telescope (**Roesler et al. 1999, Science**). The vertical bars are in parts due to contamination by geocoronal emission, in particular the green bar shows Lyman-alpha emission from the geocorona. The disks in this figure represent images of the scientific target of these observations, i.e. the emission from Jupiter's moon Io at various FUV wavelengths.

Targeting 100 km alt. resolution

Altitude resolution ~ 100 km



Payload

in-situ particle	TRL
INMS: Cold ion & neutral mass spectrometer ($M/\Delta M > 1000$): U. Bern	7 - 8
WCIMS: Cold ions & neutral f_{dist} : NASA-GSFC	7
MIMS: Light hot ions ($M < 20$, < 40 keV/q): IRAP, Toulouse	5
NOIA: Heavy hot ions ($M > 10$, < 30 keV/q): IRF, Kiruna	≥ 6
EMS: Energetic ions (20–200 keV): U. New Hampshire, USA	≥ 6
ESMIE: Electrons (~ 5 eV – 20 keV): UCL/MSSL, London, UK	≥ 6

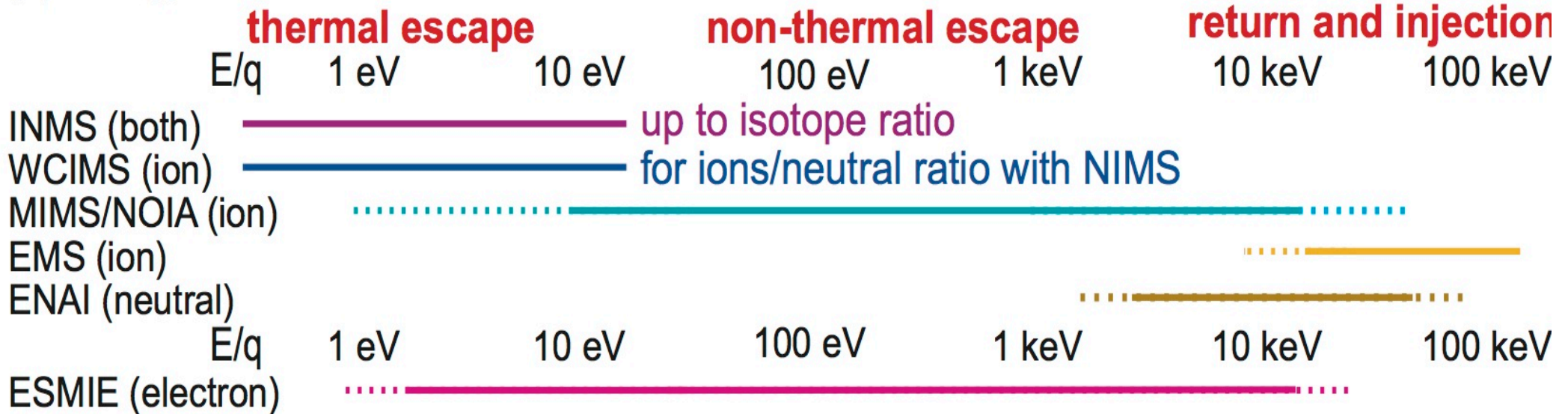
in-situ field	TRL
SLP: Sweeping Langmuir probe: e-density, spacecraft potential: BIRA-IASB, Brussels	4 - 5
MAG: Magnetic field: IWF, Graz	8
Waves (5 Hz – 20 kHz): ASCR, Prague	≥ 5
Search Coil: LPC2E, Orléans	

Line-of-sight integrated values	TRL
UVIS: UV imaging spectrometer (85 – 140 nm, 83, 58, 30 nm for H, N, O, O⁺, He, He⁺): U. Tokyo	6 - 7
AMC: Aurora monitoring camera (670 nm and 630 nm): Tohoku U.	7 - 8
ENAI: ENA imager (2–200 keV): INAF/IAPS, Rome	> 5

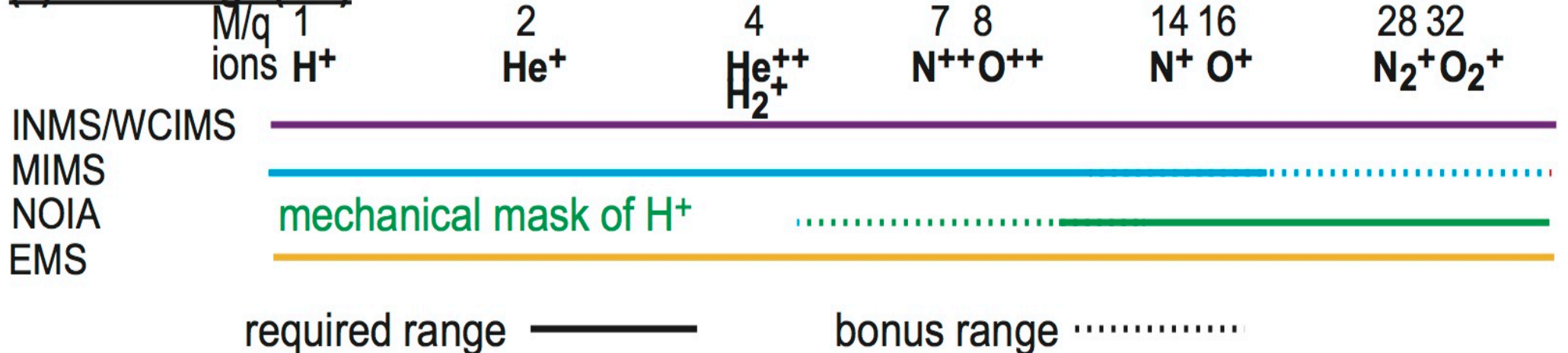
Mass coverage

Energy and mass coverage of particle instruments

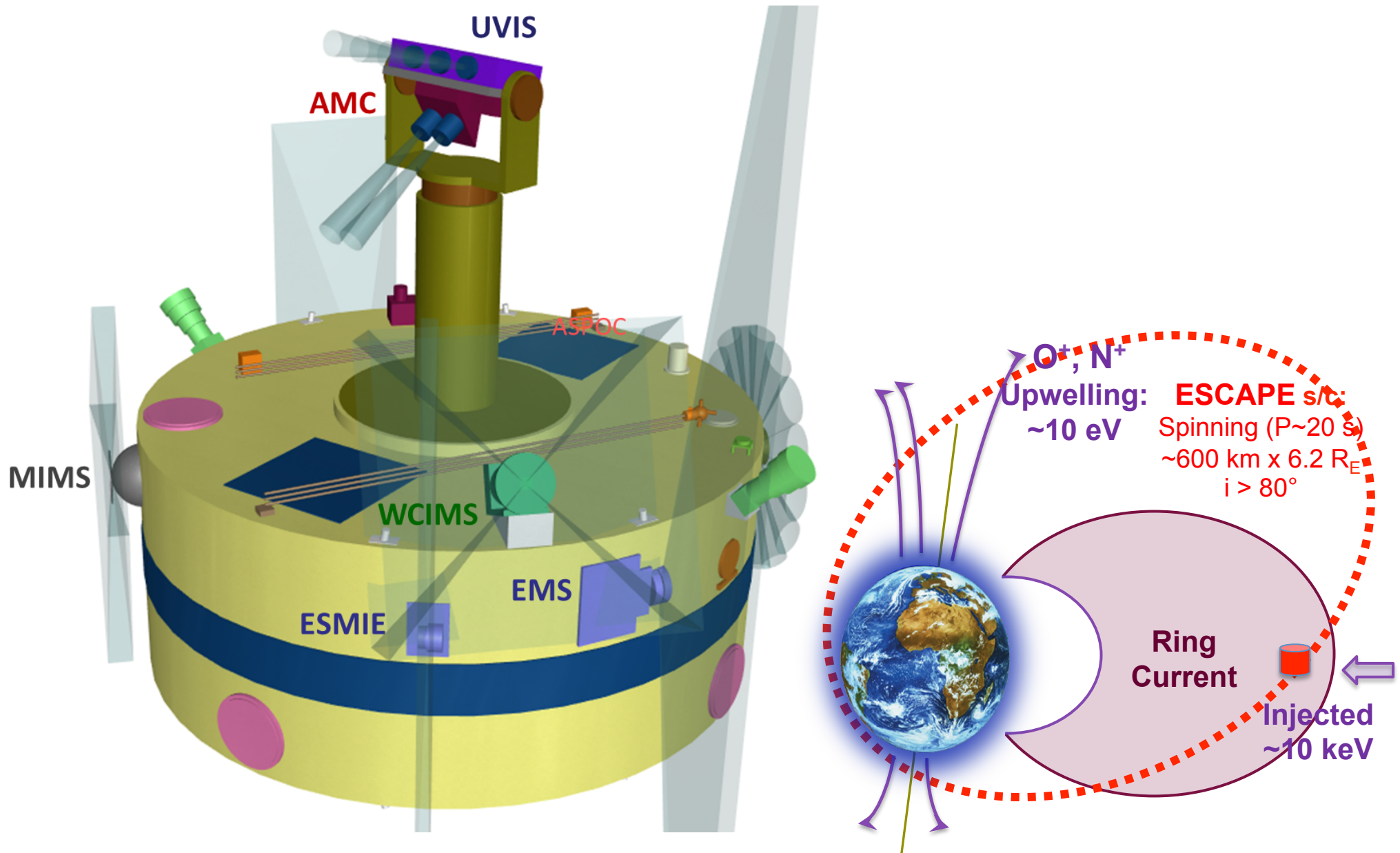
(1) Energy range



(2) Mass range (ion)



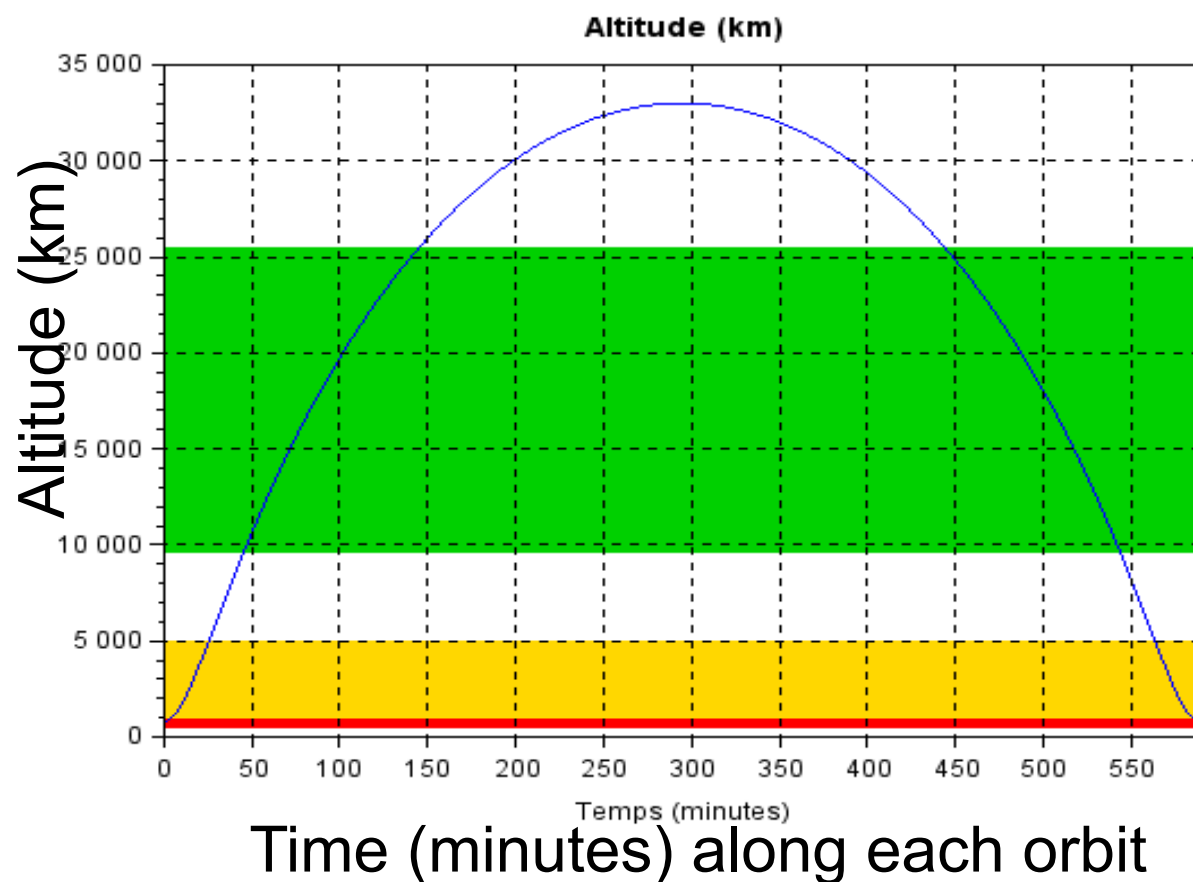
Spacecraft (spin+despun) and orbit



Orbit evolution

- Initial perigee altitude : **800 km**
- Apogee altitude : **33 000 km** (6.2 R_E geocentric distance)
- Orbital plane inclination : **90°**
- Initial latitude of the line of apsides: **85°N**
- Argument of perigee: **255°**
- **9 h 45** orbital period (2697 orbits in 3 years)
- No need for orbit maintenance manoeuvres
(unless we want to gradually change the orbit characteristics)
- **-0.21°/ day rotation** of the line of apsides in this plane (230° in 3 years)
- Slow oscillation of the perigee altitude, **between 800 and 480 km**
- Need for deorbiting at the end of mission

Region of interest



Ring current region
 ($2.5 - 5 R_E$, $\text{lat} < 45^\circ$):
 34 % per orbit,
 i.e. > 8 hours per day

Upwelling region
 (1000-5000 km, $\text{lat} > 70^\circ$):
 1.47 % of traversals,
 i.e. ~21 minutes per day

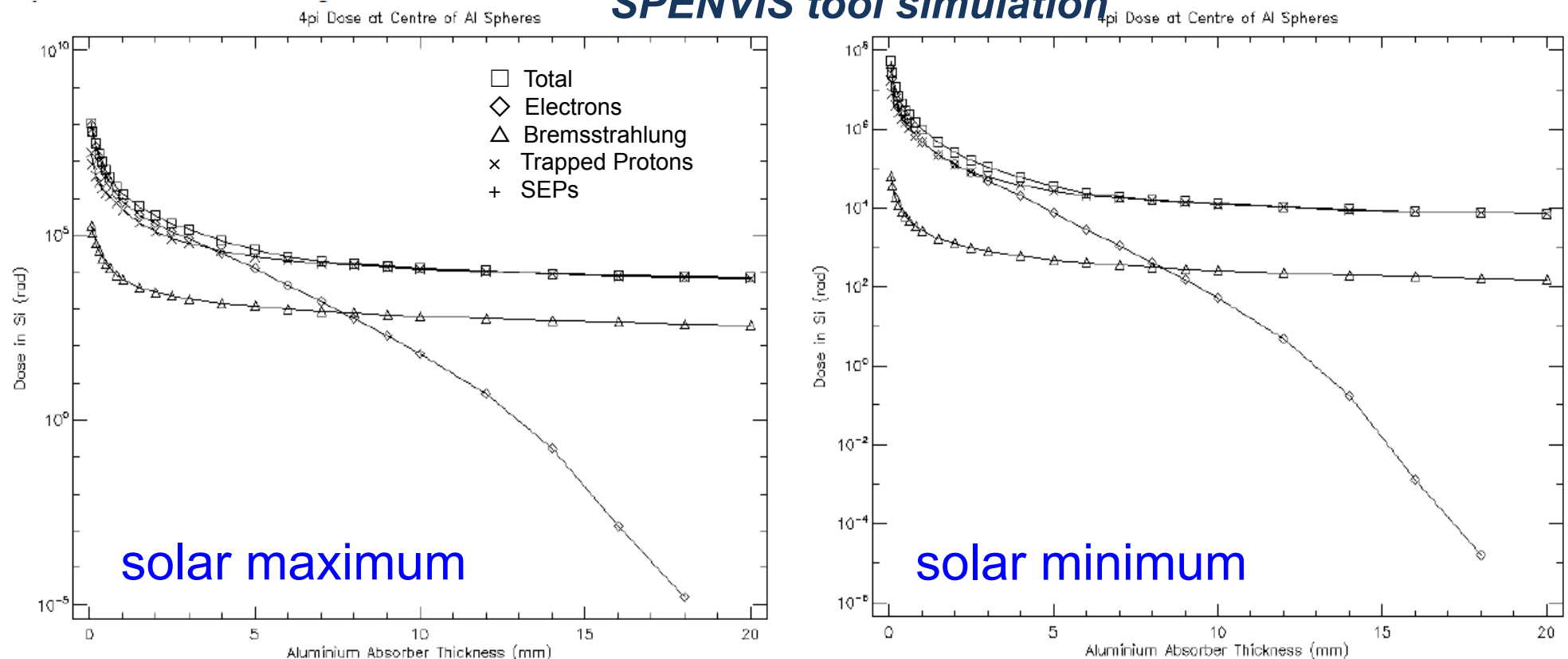
Exosphere (500-1000
 km): 1.56 % per orbit,
 i.e. ~22 minutes per day

White and green zones (> 5000 km altitude) are perfectly suited for remote sensing observations of the lower exosphere and limb.

Radiation dose

Total ionising doses after 3 years
 ~ 35–40 krad behind 5 mm of aluminium shielding

SPENVIS tool simulation



Total ionising dose in silicon (rad) as a function of the aluminium shielding thickness (mm)

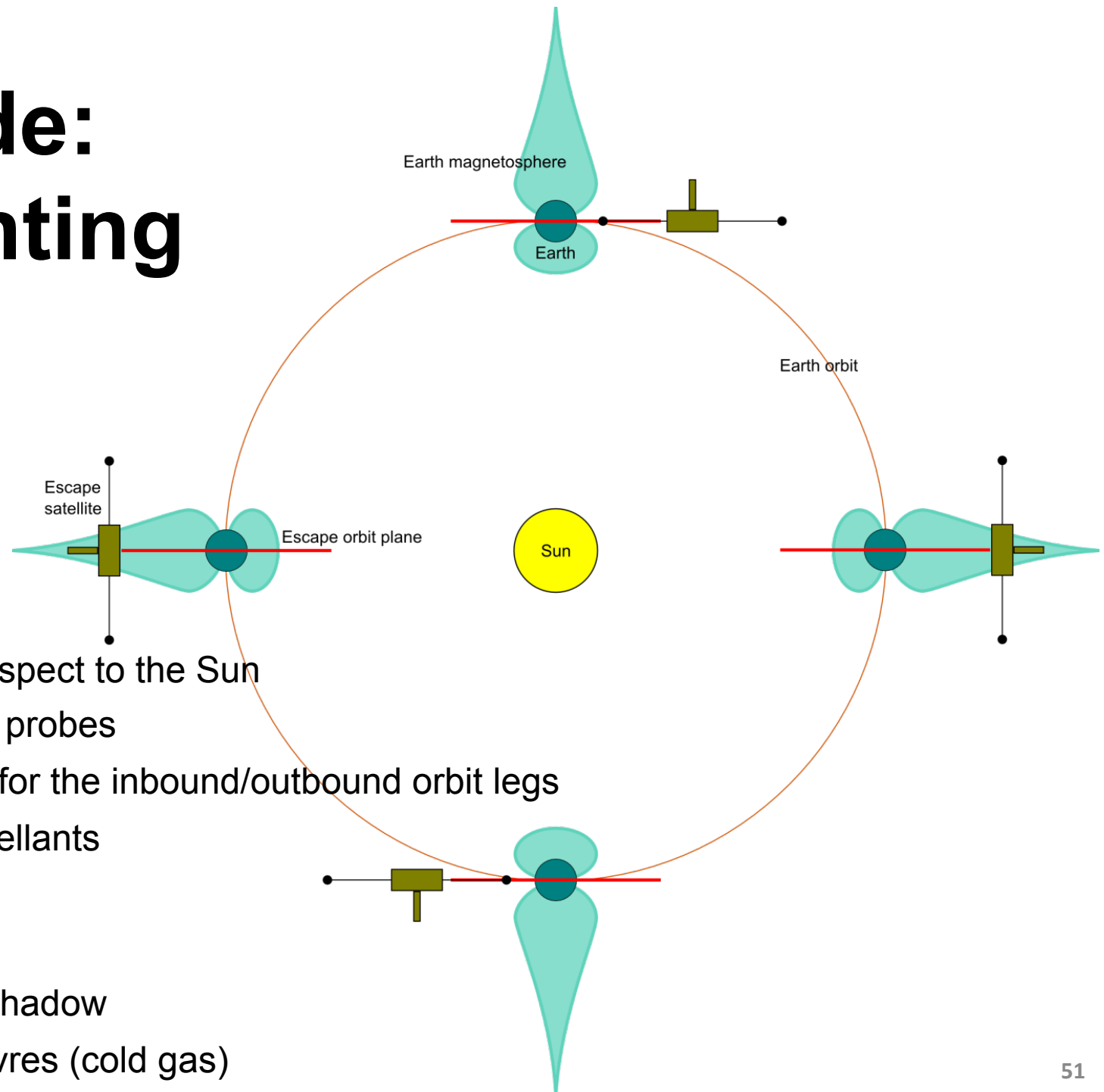
Attitude: sun-pointing

Requirements:

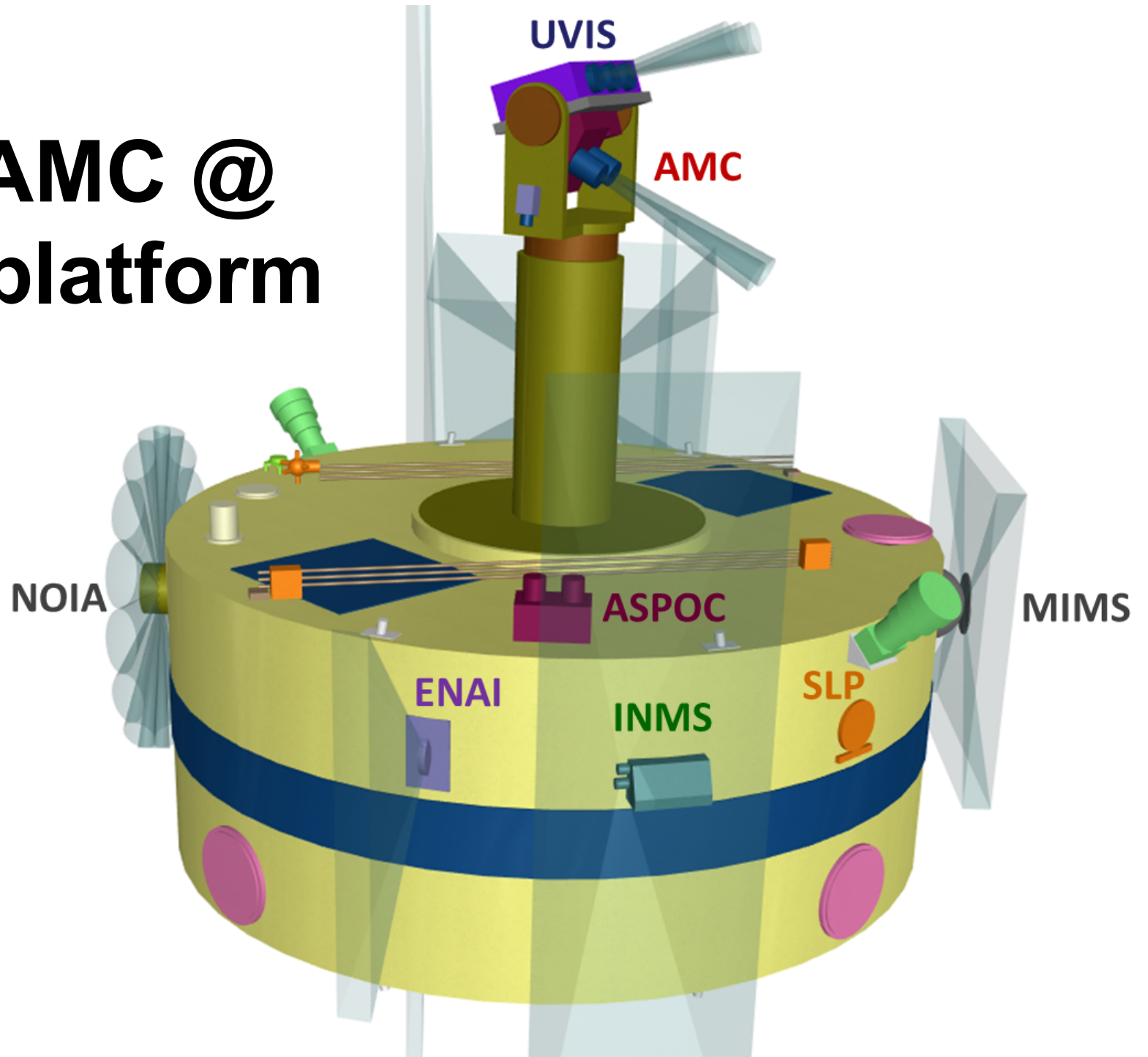
- 3 rotations / minute
- Constant attitude with respect to the Sun
- No shadows on the SLP probes
- Allow auroral zone view for the inbound/outbound orbit legs
- No nitrogen or CH₄ propellants

Resulting Design:

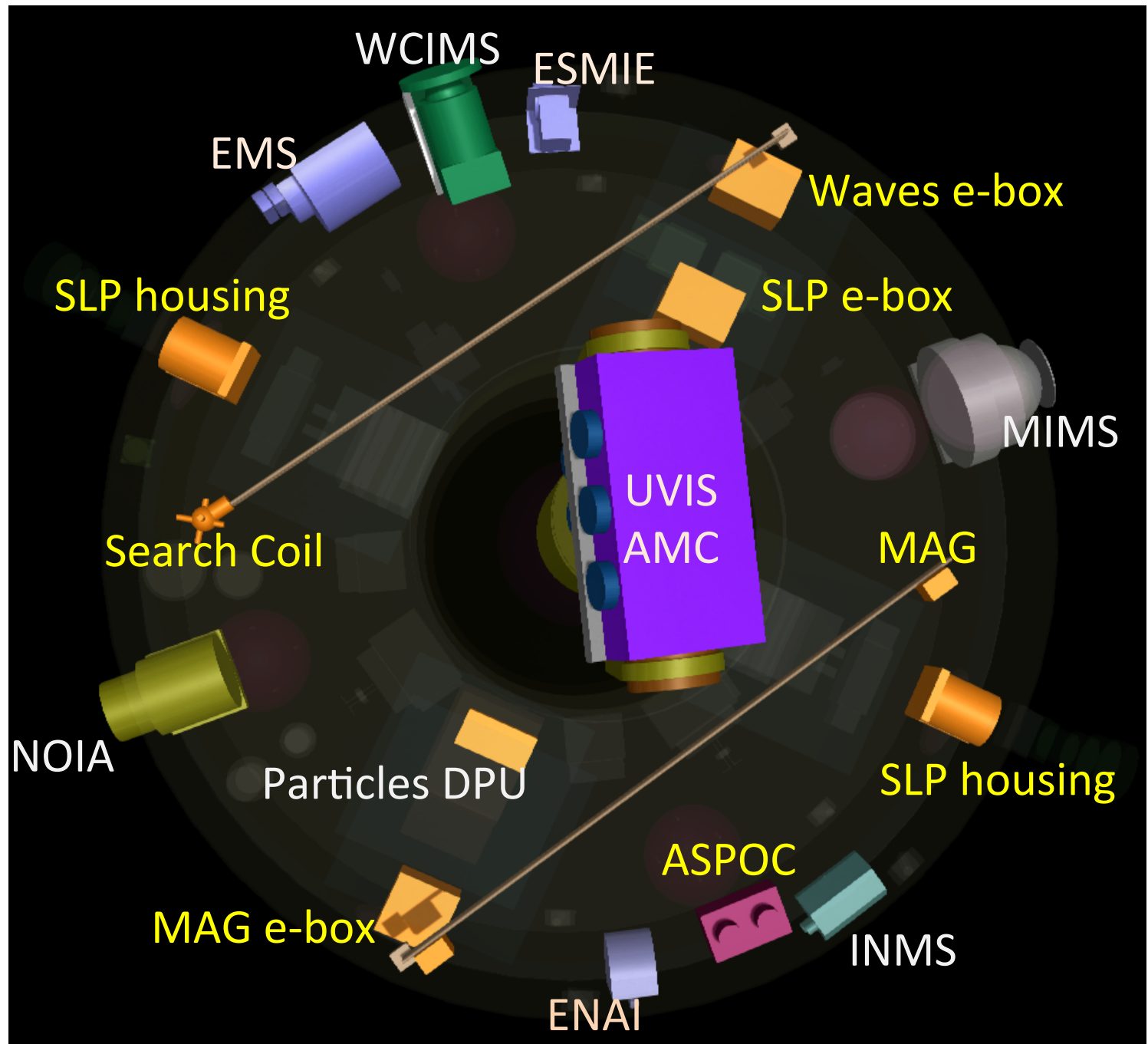
- Sun-pointing spin axis
- Despun platform in the shadow
- 1°/ day attitude manoeuvres (cold gas)

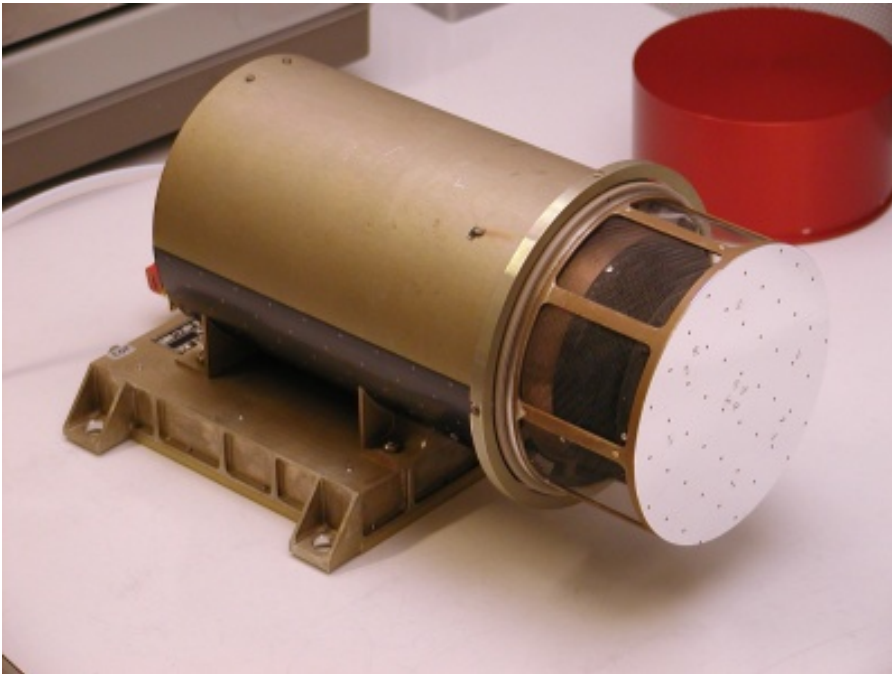


UVIS + AMC @ despun platform



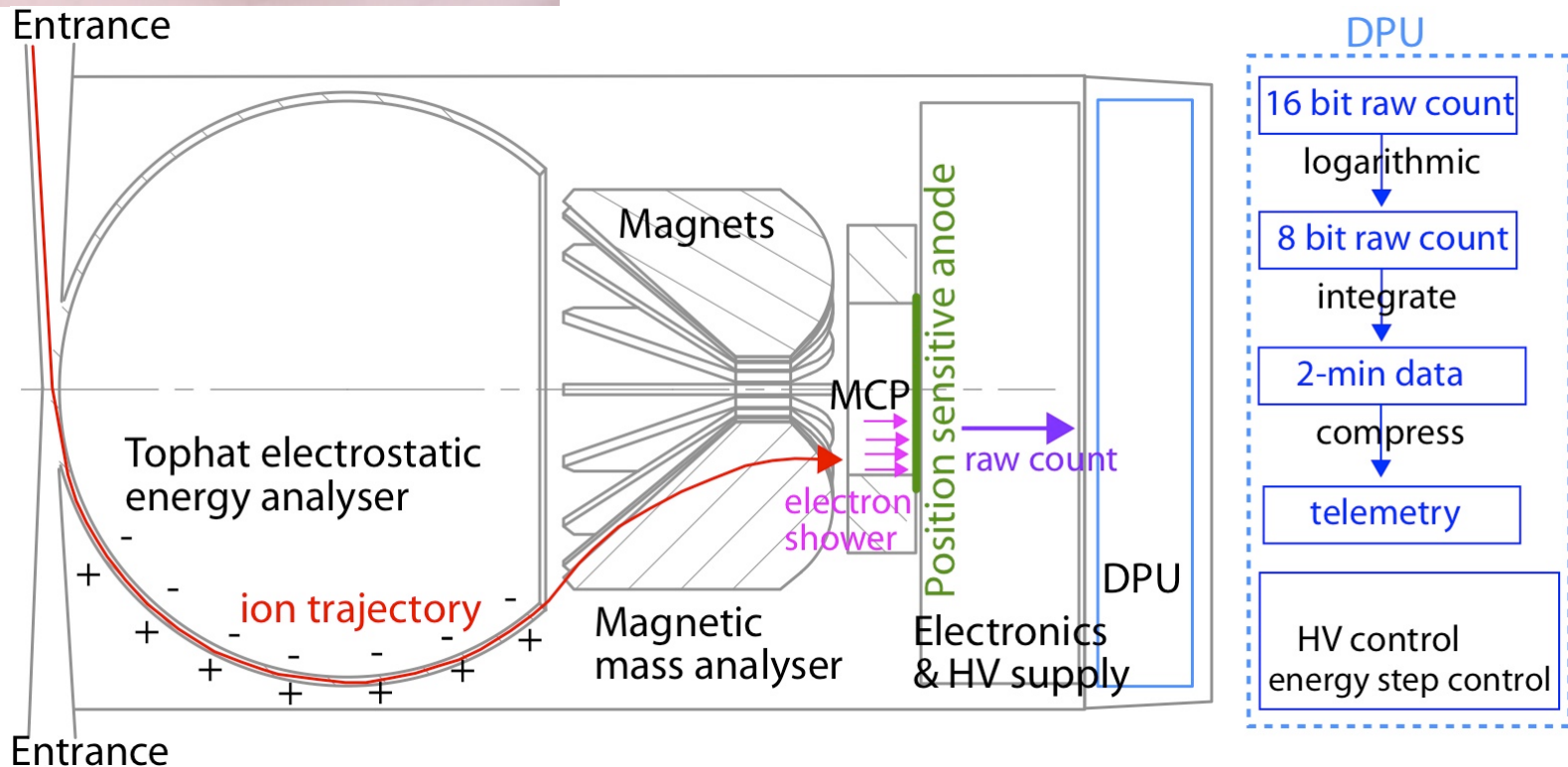
Accommodation



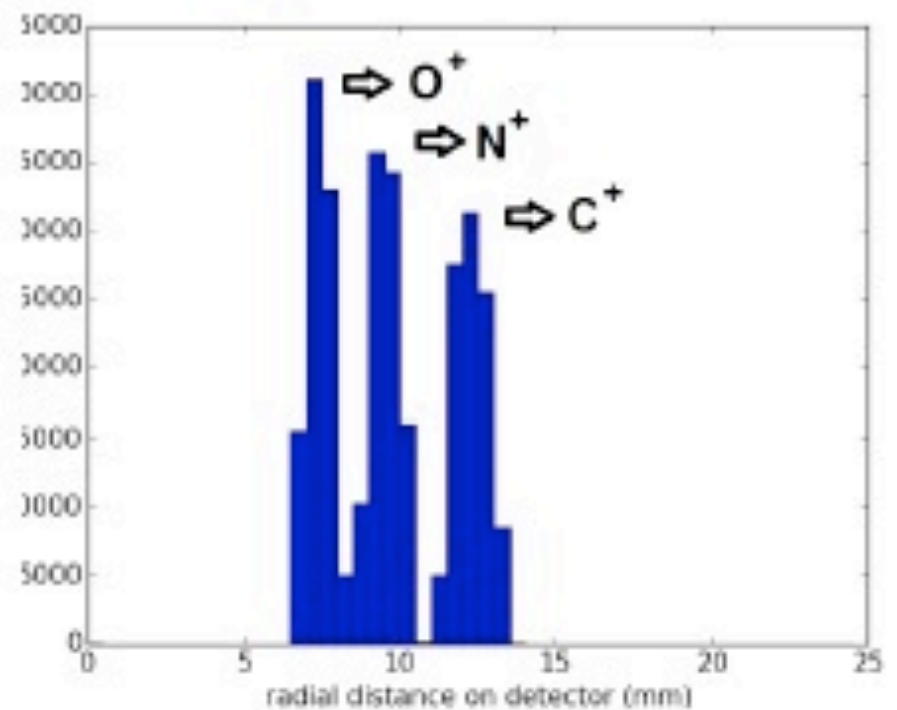
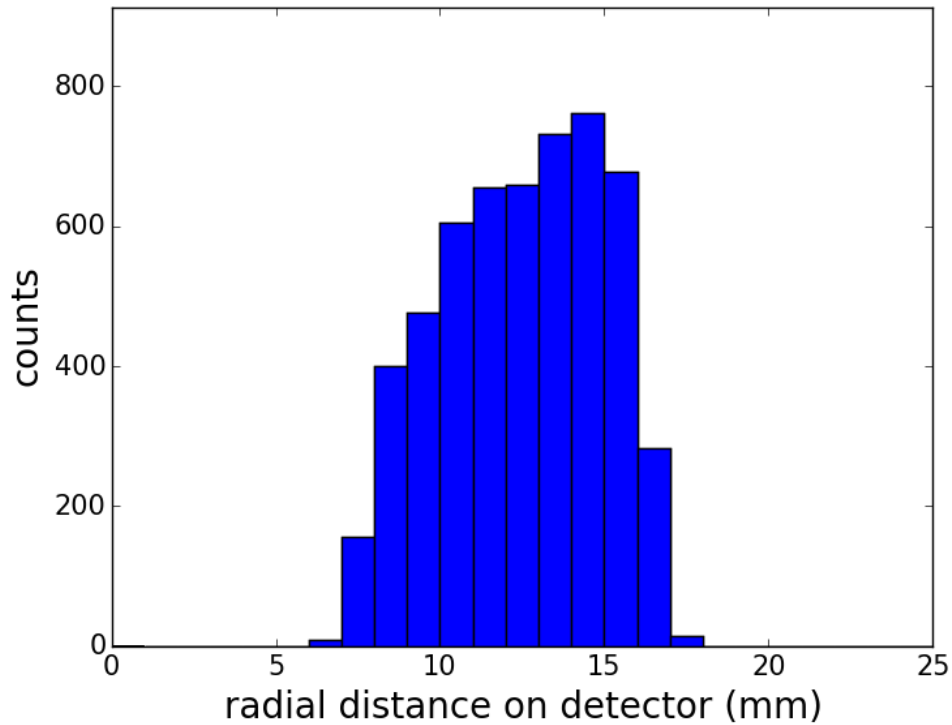


IMA/ICA → NOIA

Will be 4 cm longer and 6 cm wider than this, but entrance is narrower



estimated resolution

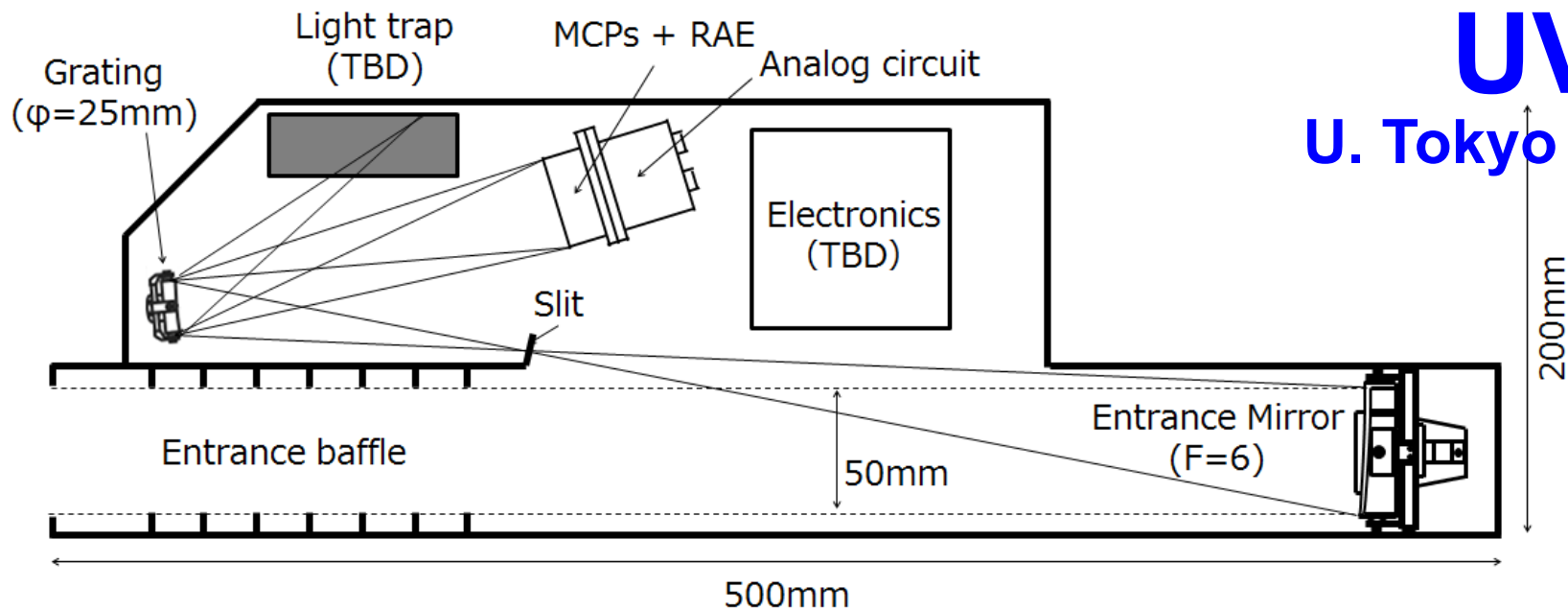


By Georgios Nicolaou

END

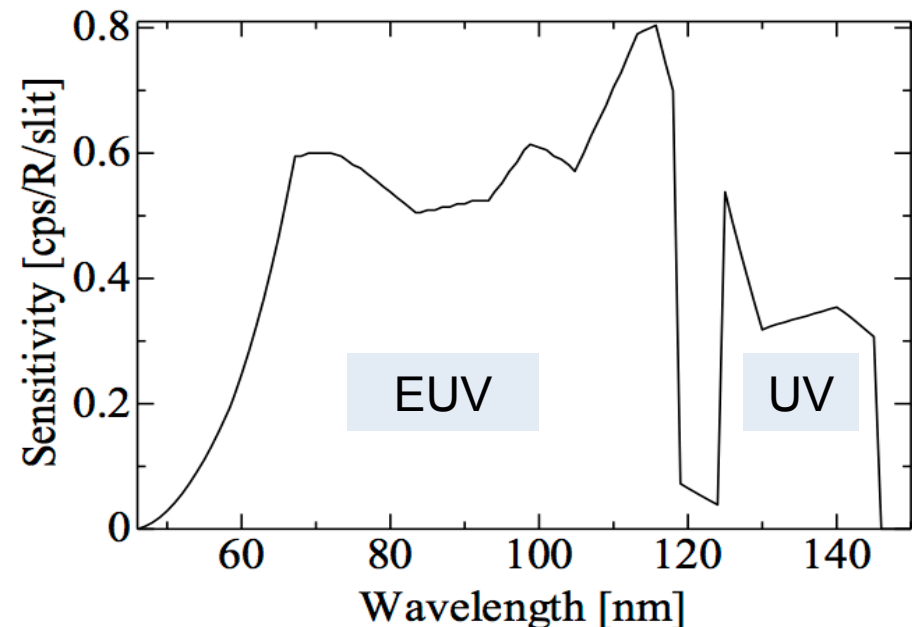
UVIS

U. Tokyo (Yoshikawa)



Size: 100 x 200 x 500 mm

- 20 slits of $0.1^\circ \times 1^\circ$
- 85–140 nm for H, N, O; and at 83 nm (O^+), 58 nm (He), 30 nm (He^+).
- Size: 10 cm x 20 cm x 50 cm with $\text{Ø} = 5$ cm mirror
- Mass: 7 kg (after 7 mm Al)

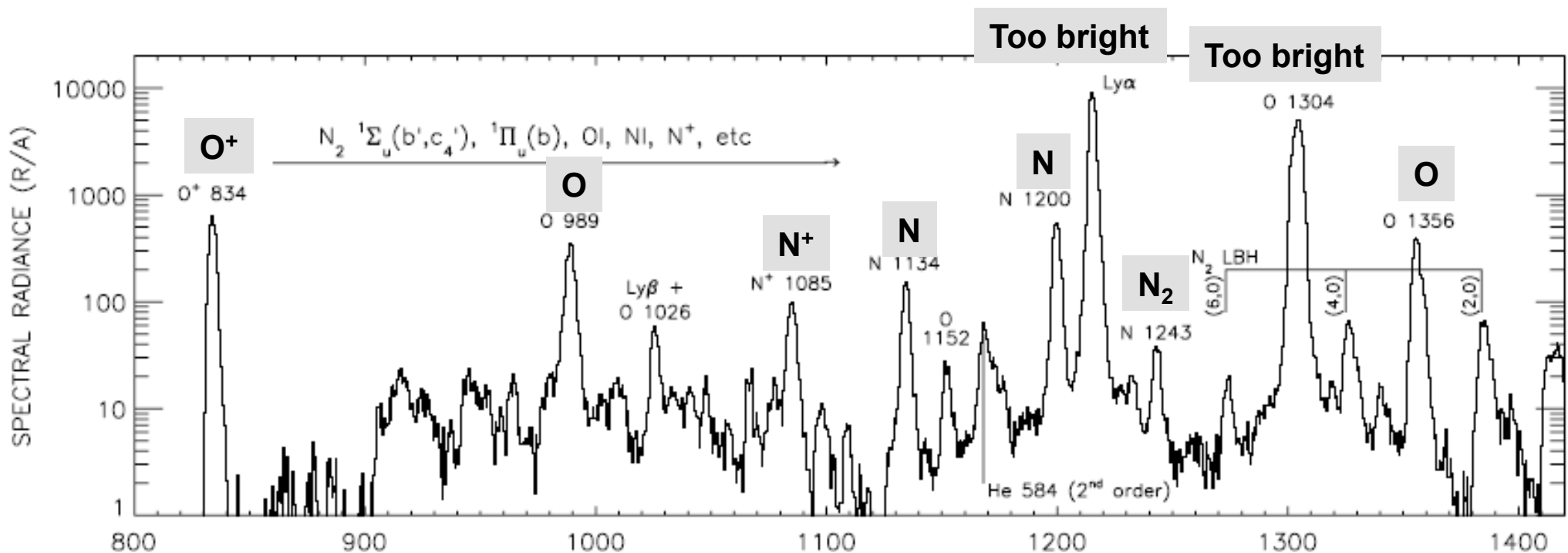


Science of UVIS

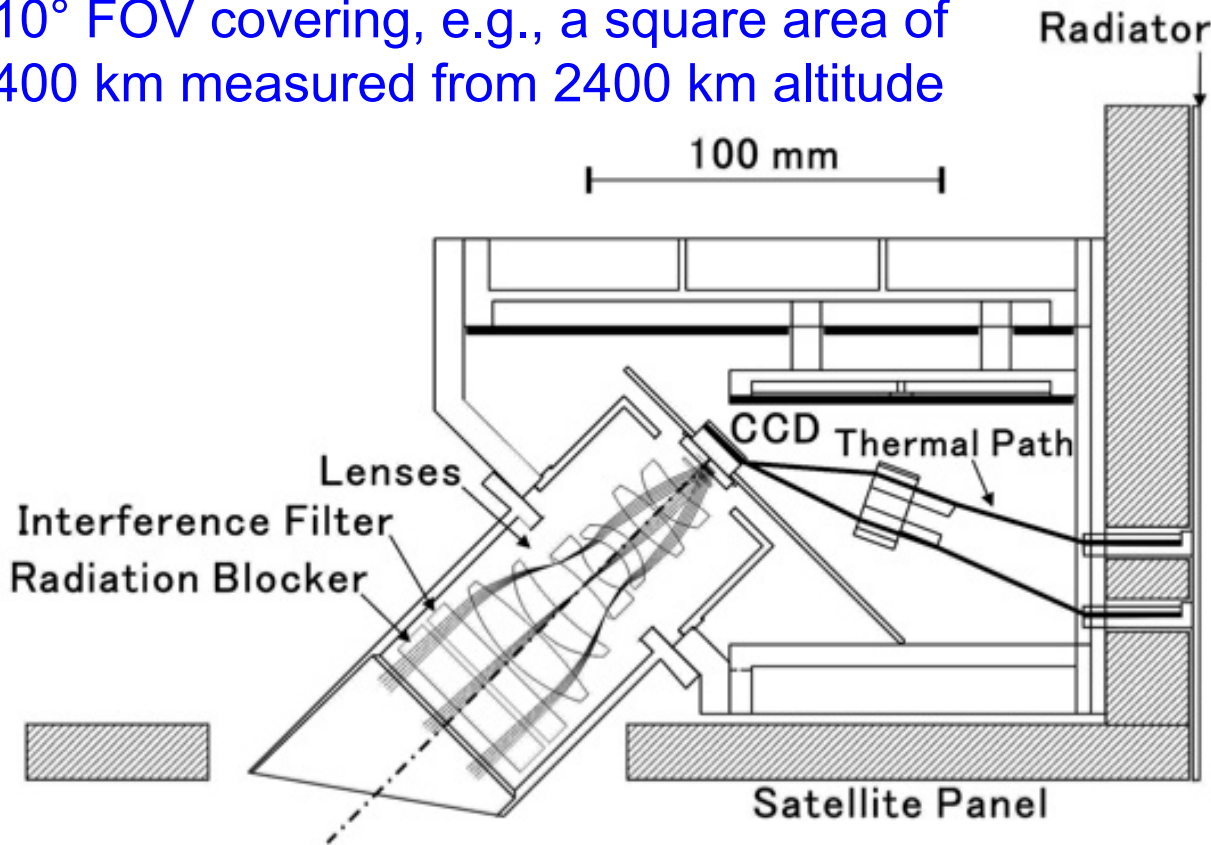
skip

Density altitude profiles (+ variability) of neutrals and ions in the exosphere (O, N, N₂, H, He).

Will measure in coordination with other instruments



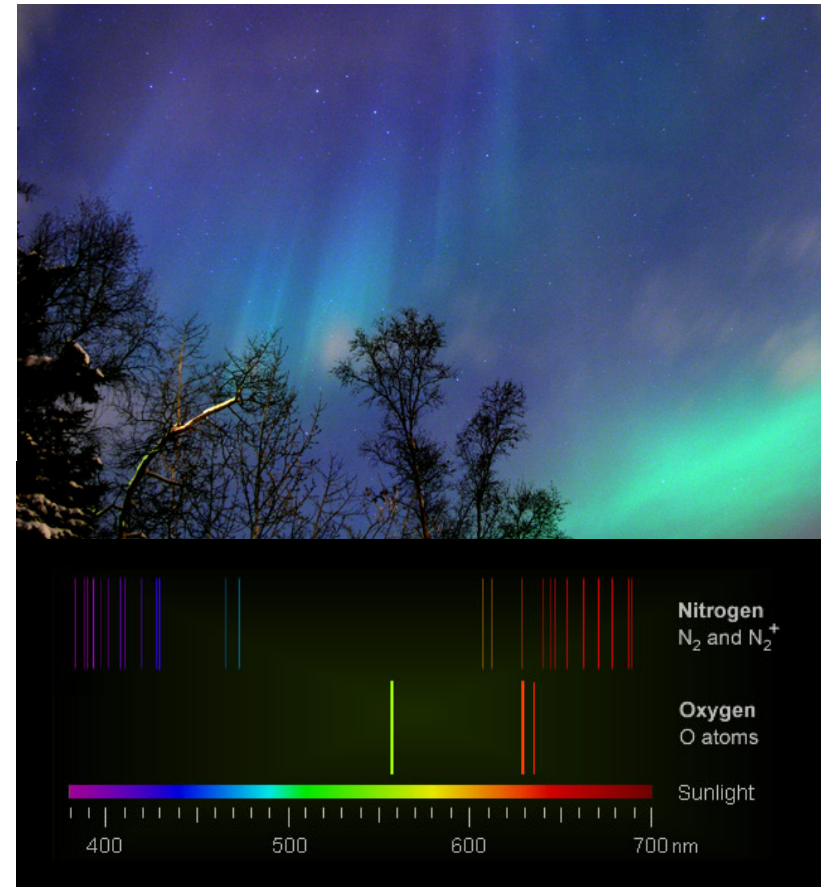
10° FOV covering, e.g., a square area of 400 km measured from 2400 km altitude



AMC

Tohoku U. (Sakanoi)

- O : 630 nm (high altitude)
- N₂ : 670 nm (very low altitude)
- O : 558 nm (relatively low altitude, optional)



AI: space-proof CCD instead of commercial comp.

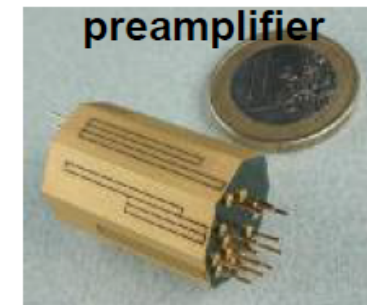
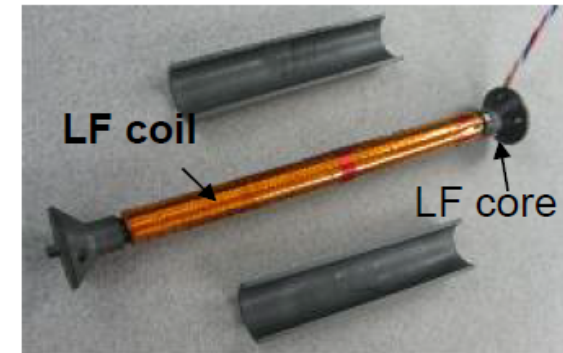
Visible

Search Coil (wave)

Col: Kanazawa U. (Yagitani)

- **Sensor (ELF-VLF)**
 - 3 mono band ELF-VLF antennas [5 Hz – 20 kHz]
- **Preamplifier**

3 channels (3 ELF-VLF).
Miniaturized preamplifier integrated inside the sensor's foot.
- **Sensor's foot**
 - orthogonal assembly of the 3 antennas
 - mechanical interface



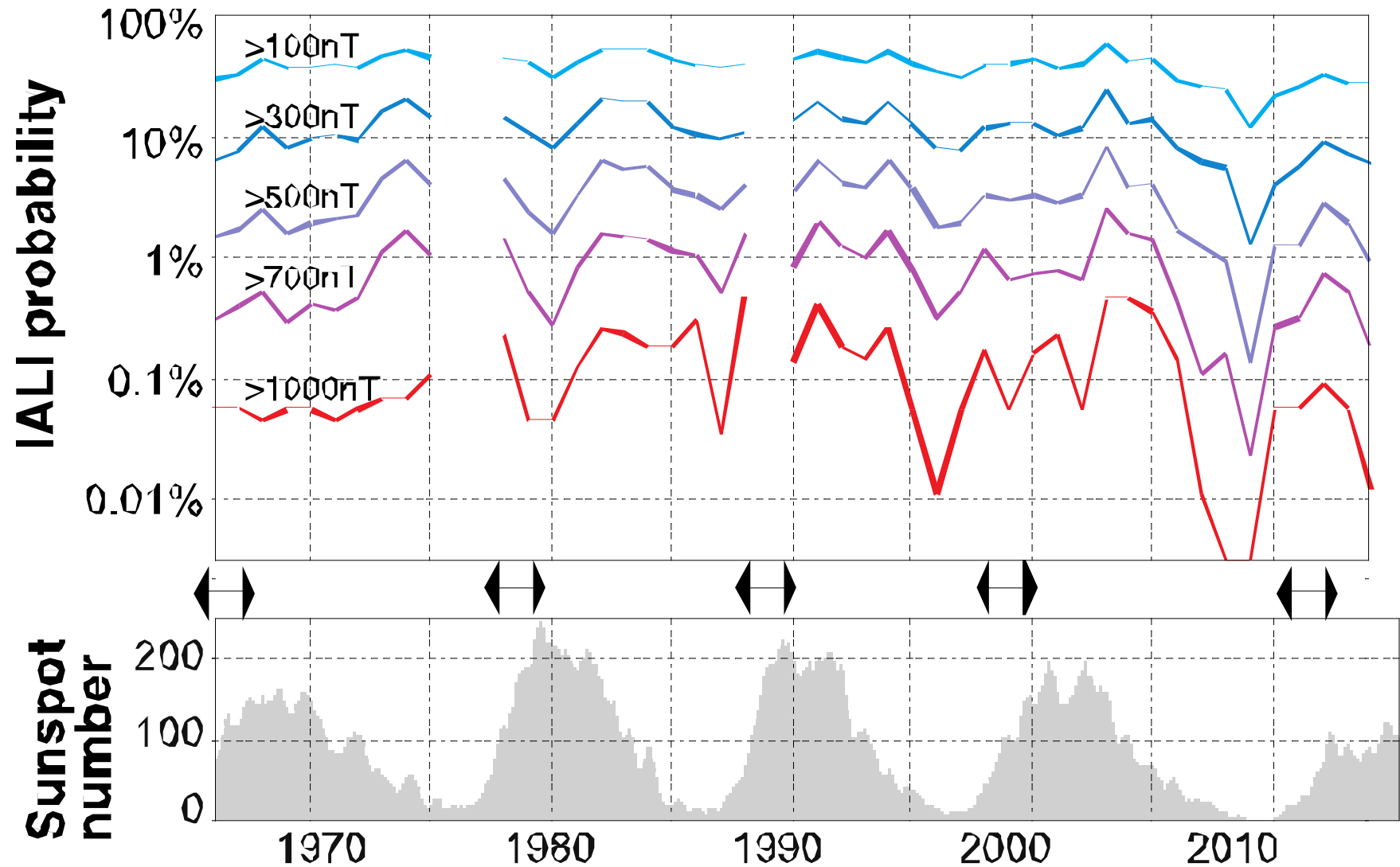
S/C specification

skip

orbital period	~10 hr
attitude control	Spin (22-26 sec)
attitude reference	Sun-pointing
control method	cold gas
time resolution	≤ 1 min
angular resolution	//, \perp , and anti-//
telemetry	~ 200 kbps
life time	3 year

Are 3 years sufficient?

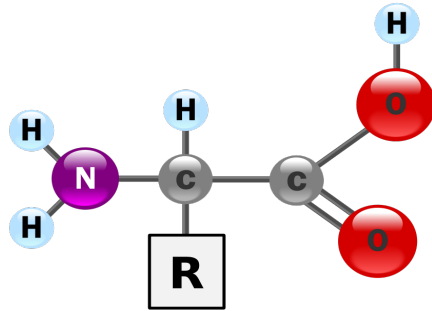
skip



We expect the inclining phase during 2030-2033

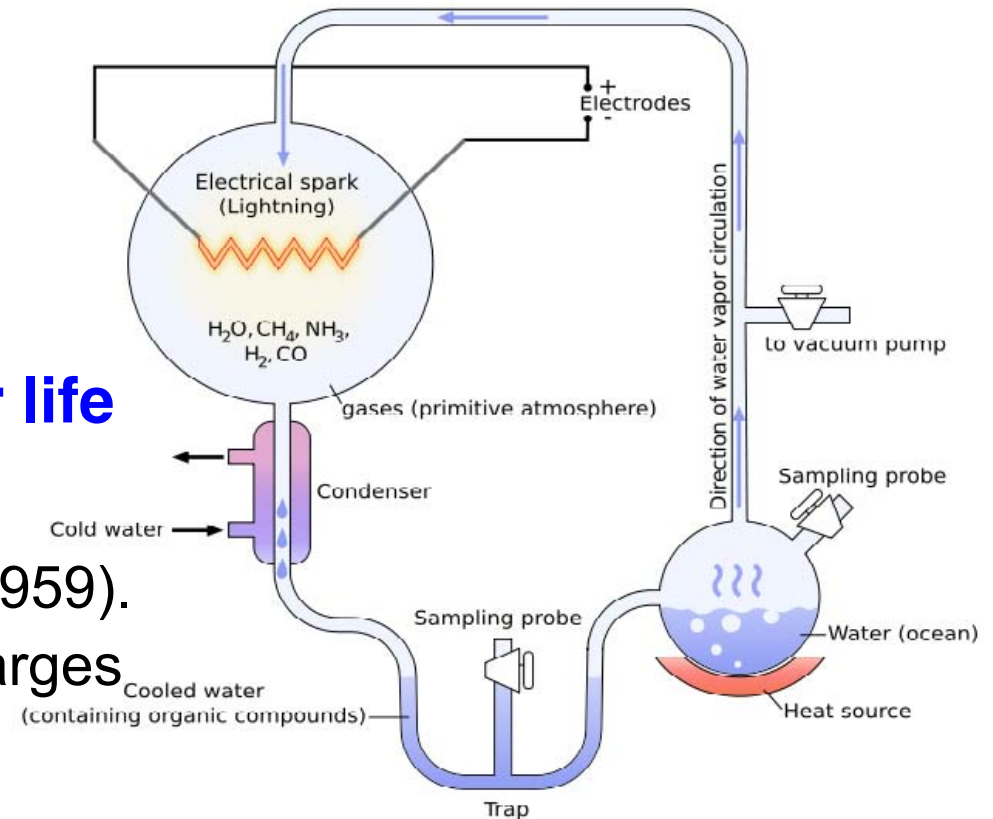
Why nitrogen? 2. Important

skip



Nitrogen is an essential element for life

Miller's experiment (Miller and Urey, 1959).
Pre-biotic type atmosphere + discharges
⇒ formation of **amino-acids** !



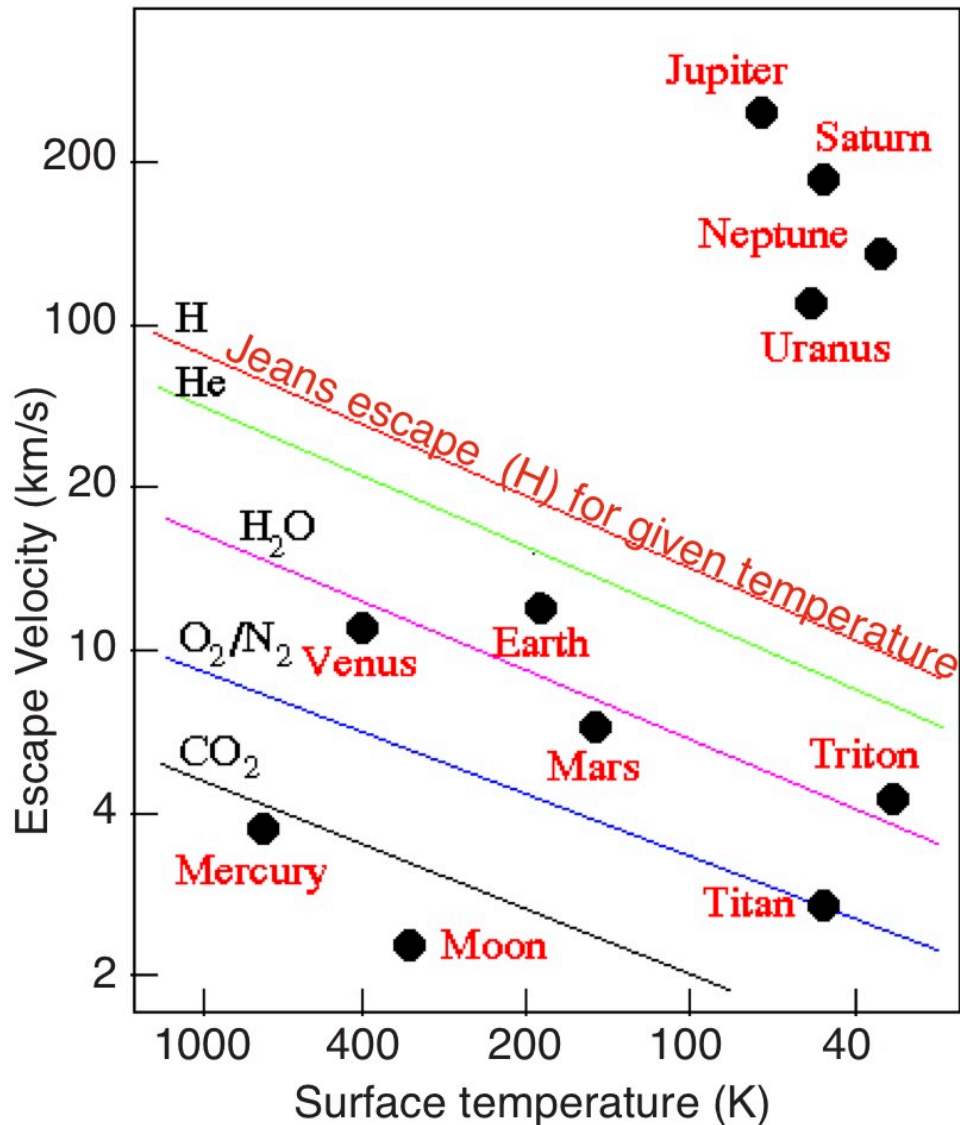
The result depends on the **oxidation state of N**

- reduced form (NH₃)
- neutral form (N₂)
- oxidized form (NO_x)

Nitrogen is missing on Mars

skip

Jeans escape does not explain it.



Planetary formation does not easily explain it.

Condensation temperature of $T_C(N_2) \sim T_C(CO) \ll T_C(CO_2) \sim T_C(NH_3)$

indicates that condensation of N and C most likely occurred in the form of N_2-CO pair ($\sim 30^\circ K$) or NH_3-CO_2 pair ($50 \sim 90^\circ K$).

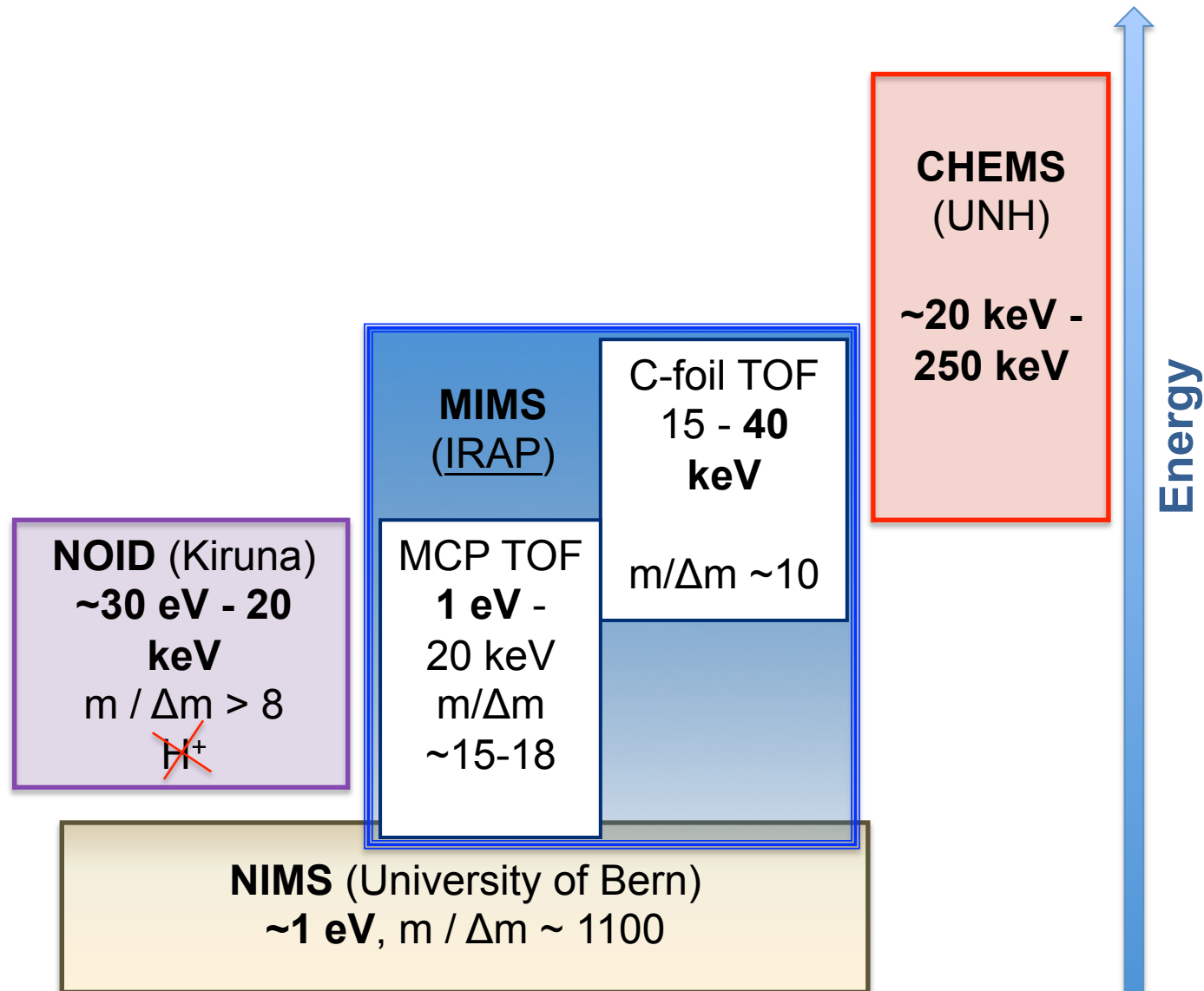
Therefore, N_2 content should be Mars > Earth > Venus.

N_2 -delivery model does not easily explain it.

It should deliver on Mars too

Omit

Mass and energy range



Roles of support instruments

skip

Magnetometer (shortest boom will do the work)

Pitch angle information and large-scale current system

Langmuir Probe (shortest boom will do the work)

Spacecraft potential for accurate ion energy measurements.

Electron detector

Identifying the region in terms of plasma region

Photoelectron information gives connectivity to the ionosphere

Aurora monitoring camera (baseline/optional ?)

Monitoring the ionospheric (ion source) condition

Wave package

Modes of waves associated with energization of ions

Energetic Neutral Atoms

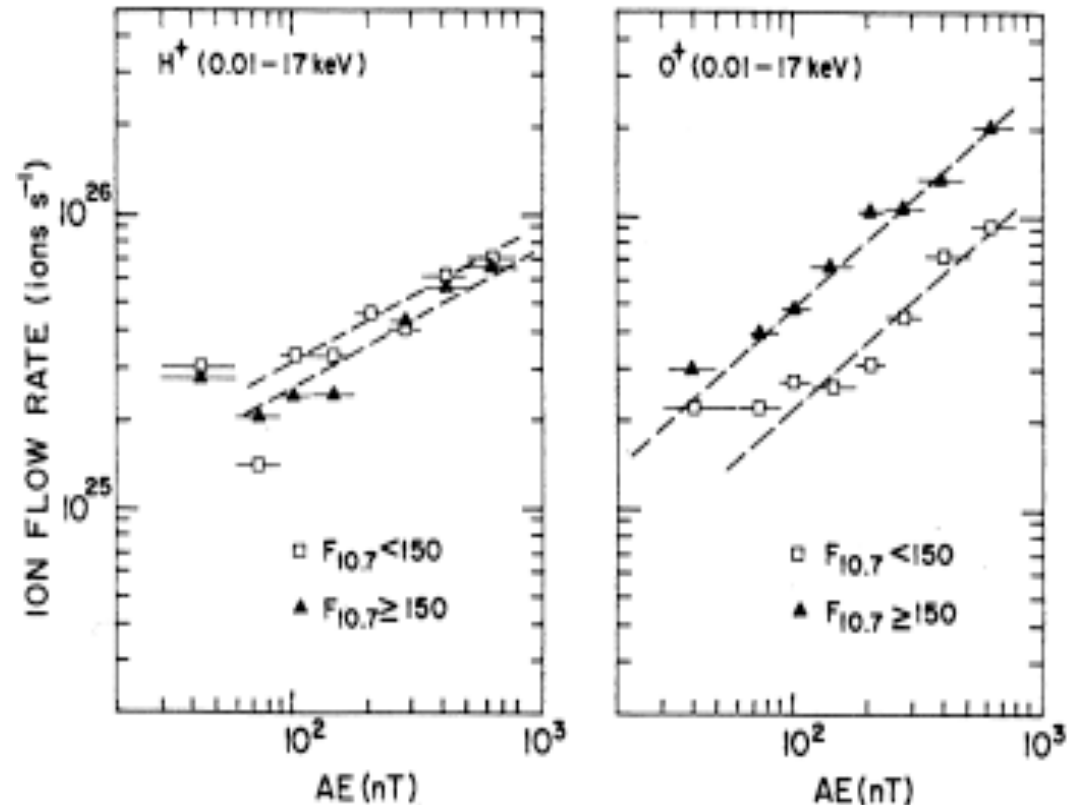
In-situ monitoring of substorm activity

Why nitrogen? 3. Amount

> 10 keV (Peterson et al., 2002)

Altitude (km)	H ⁺	O ⁺ (inc N ⁺)
50000 km (Polar)	1 · 10 ²⁵	1.5 · 10 ²⁵
20000 km (DE-1)	1 · 10 ²⁵	7 · 10 ²⁴
8000 km (Polar)	3 · 10 ²⁴	3 · 10 ²⁴
9000 km (Akebono)	2 · 10 ²⁵	2 · 10 ²⁴
3000 km (FAST)	-	2 · 10 ²⁴

(Peterson et al., 2006)



Polar/TIMAS
(Peterson, 2002)

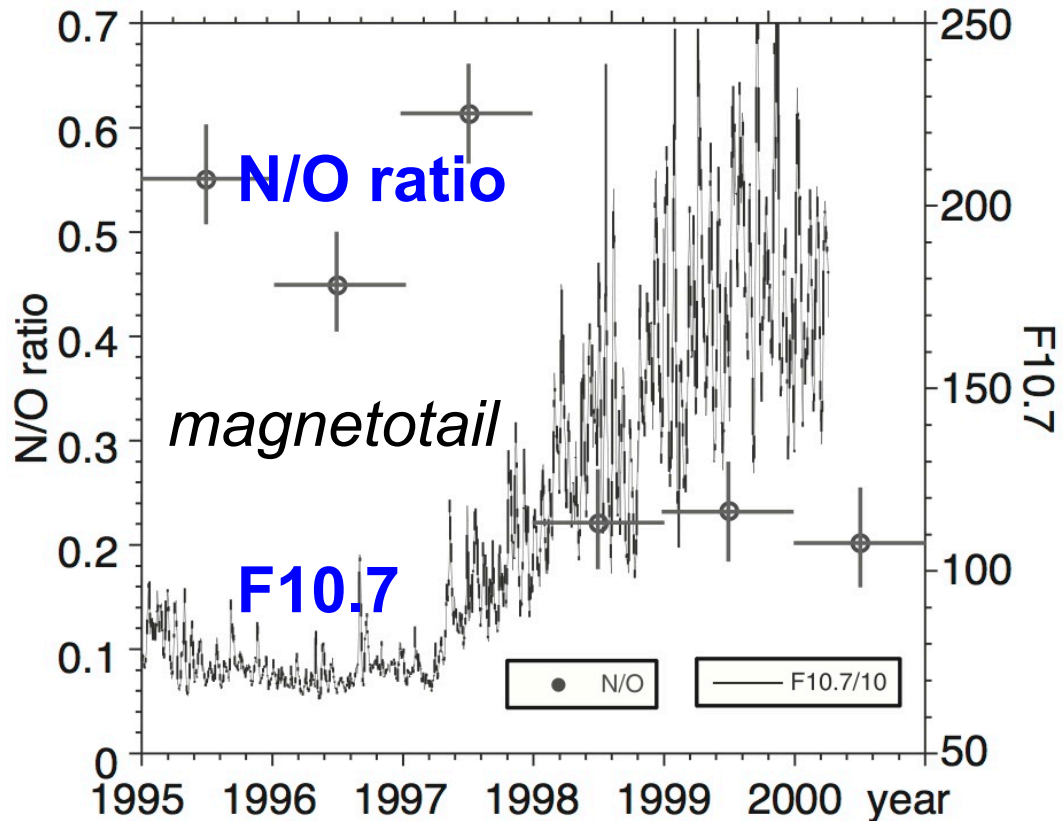
Omit

Why N^+ and N_2^+ ?

some puzzle

skip

WIND/PHA observations at 8-38 RE (10-210 keV)



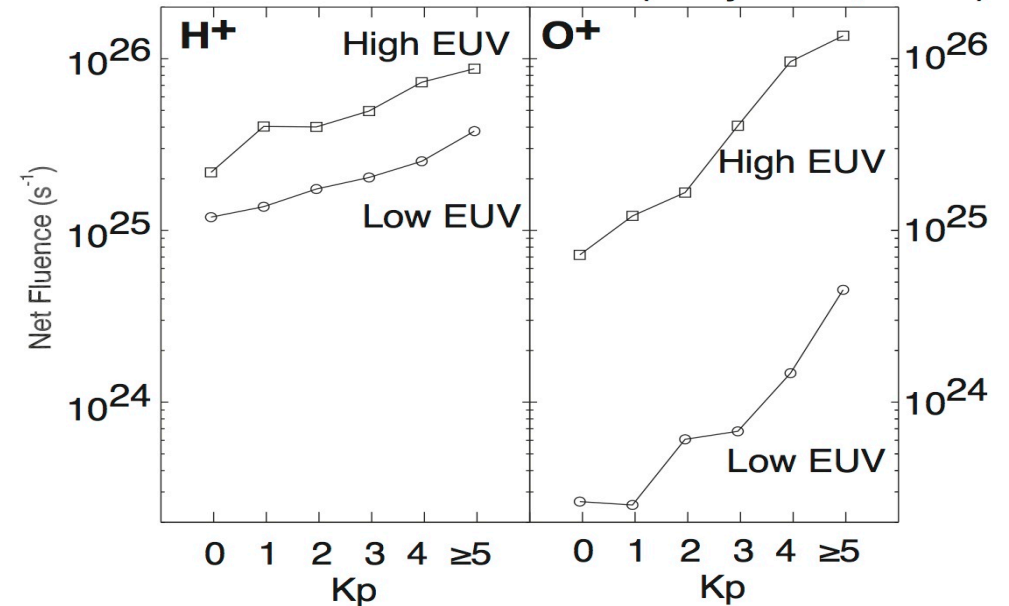
(Mall et al., 2002)

N/O ratio in **anti-phase** with activity

O/H ratio also increases with activity

O^+/H^+ increase to $\gg 1$

Akebono cold ion statistics (Cully et al., 2003)



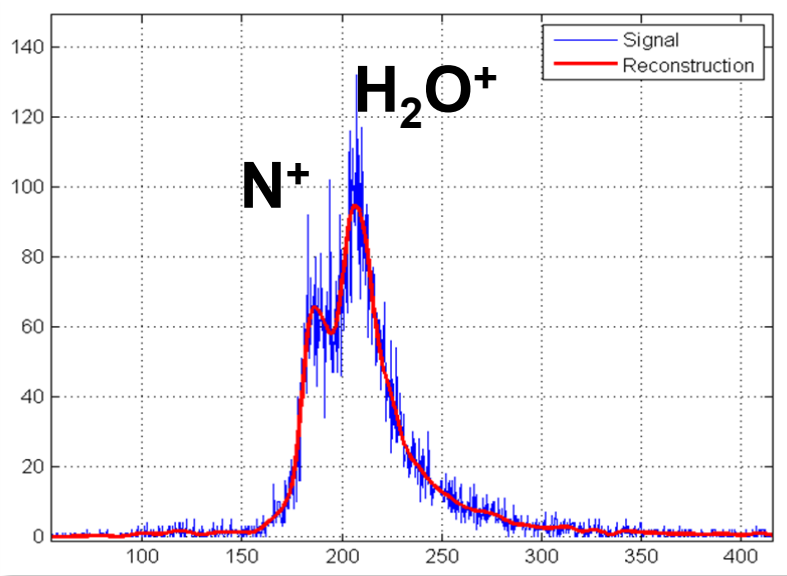
energy < 20 eV, high EUV: $F_{10.7} > 150$

The same trend for 0.02-30 keV
(Polar/TIMAS, Peterson et al., 2002)

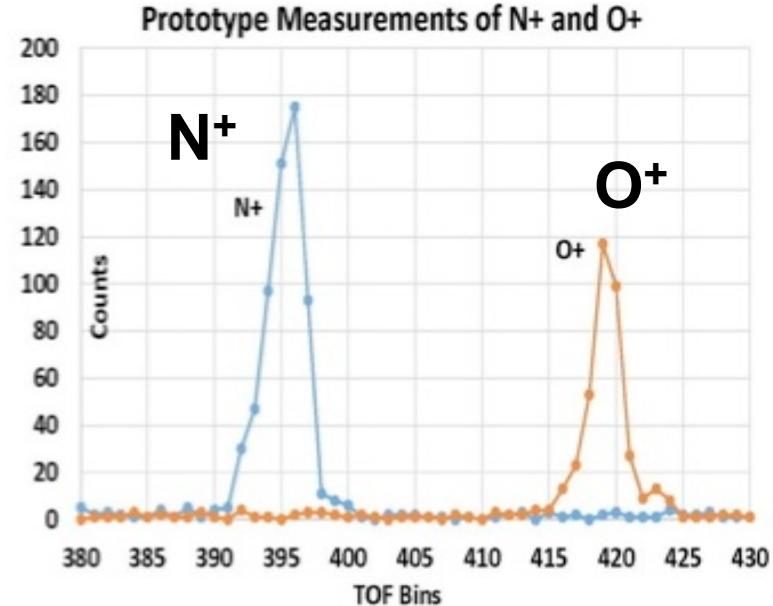
Why nitrogen? possible to det

skip

Hot ion (lab calibration)

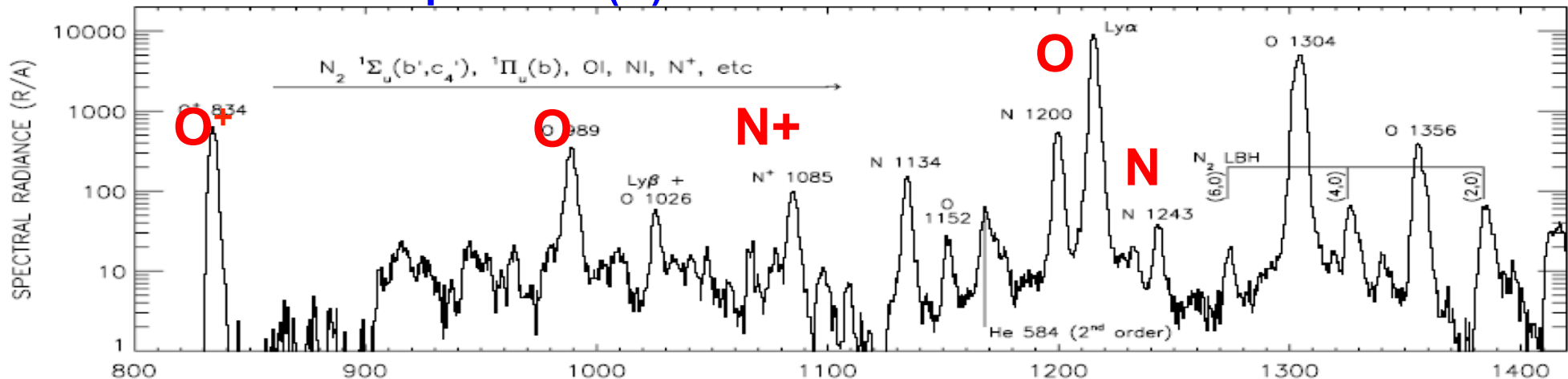


Cold ion (lab calibration)



69

UV Spectrum (Å)



69

Our knowledge on Earth's N⁺ behavior is poor

- (a) Dependence on geomagnetic activities is **larger for N⁺ than O⁺** for both <25 eV (Yau et al., 1993) and > 30 keV (Hamilton et al., 1988).
- (b) **N⁺/O⁺** ratio varies from <0.1 (quiet time) to ≈ 1 (large storm). What we call **O⁺** is normally a mixture of **N⁺** and **O⁺**. This also applies to **O⁺⁺**.
- (c) [CNO group]⁺ at <10 keV range is **abundant in the magnetosphere**.
- (d) Ionization altitude of **N** (eventually **N₂**) is likely higher than for **O** in the ionosphere (when **O⁺** is starting to be heated, majority of **N** is still neutral).
- (e) **N/O ratio at Mars** (and C/O ratio at Moon) **are extremely low** compared to the other planets.
- (f) Molecular N₂ was detected in Martian soil and comet, but the ratio was very low.
- (g) Isotope ratio (e.g., **¹⁵N/¹⁴N**) is different between different planets/comets.

One thing clear is that O⁺ behavior and N⁺ behavior are completely different!

Omit

baseline payload

Measurement	SI (PI institute)	Required ability to measure
light hot ions:	MIMS (IRAP)	H+, He ⁺⁺ , He ⁺ , O ⁺⁺ , N ⁺ , O ⁺ (10 eV - 20 keV),
heavy hot ions:	NOID (IRF)	N ⁺ , O ⁺ , N ²⁺ (10 eV - 20 keV/q),
cold ions:	NIMS (UBern)	H+, He ⁺⁺ , He ⁺ , N ⁺⁺ , O ⁺⁺ , N ⁺ , O ⁺ , N ²⁺ , O ²⁺ (< 10 eV),
energetic ions:	CHEMS (UNH)	H+, He ⁺⁺ , He ⁺ , O ⁺⁺ , N ⁺ , O ⁺ (20-200 keV/q),
SC potential:	SLP-IS (BIRA-IASB)	1 V accuracy, every spin
UV/visible emission:	NUVO (LATMOS)	91 nm (N ⁺), 108 nm (N ⁺), 391 nm (N ²⁺), 428 nm (N ²⁺)
magnetic field:	MAG (IWF)	-5000 - +5000 nT,
electrons:	PEACE (MSSL)	10 eV - 10 keV,

Omit

optional payload

Measurement	SI (PI institute)	Required ability to measure
IR emission:		
airglow/aurora emission	CAAC (TohokuU)	two of auroral emission
cold ions and neutrals:	CINMS (NASA/ GSFC)	H+, He++, He+, N++, O++, N+, O+, N2+, O2+, N, O
wave analyser	WAVES (ASCR/IAP)	10 Hz – 1 KHz,
waves detector		together with WAVES
ENA		

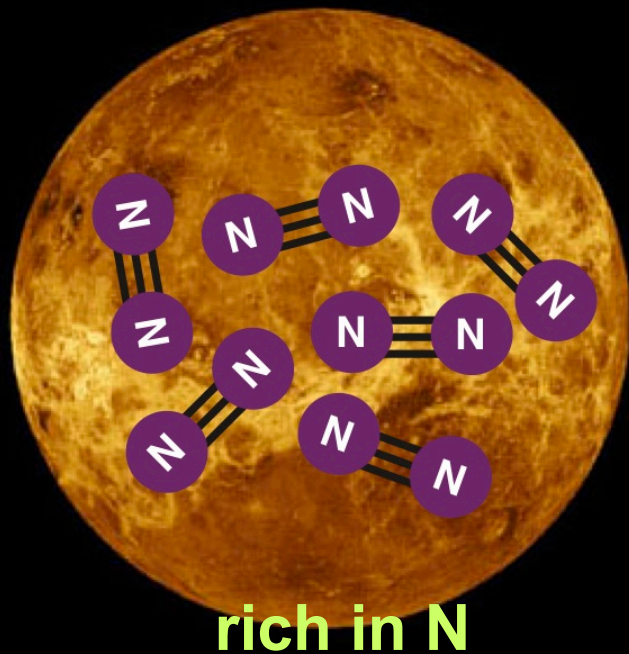
Omit

Open questions on Nitrogen

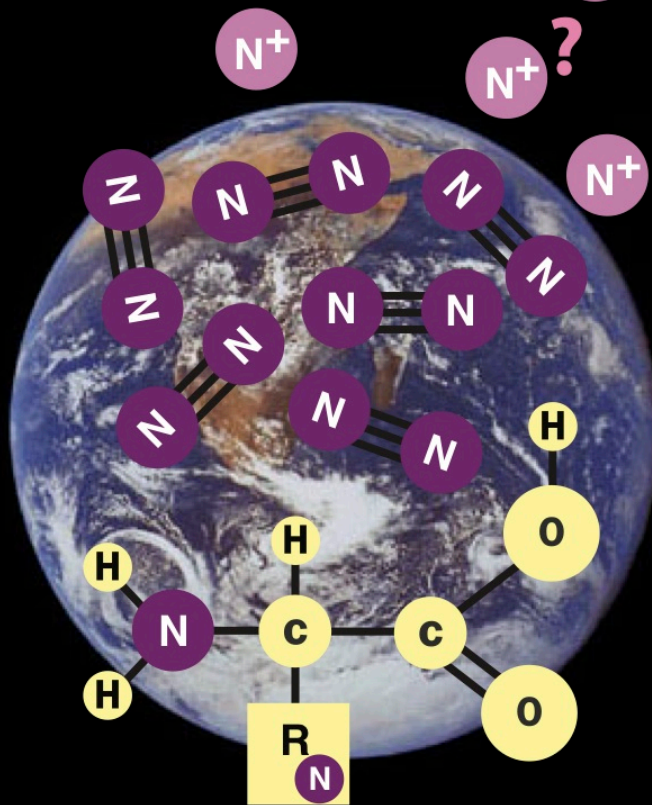
1. Nitrogen form @ ancient Earth?

2. Why nitrogen @ Mars

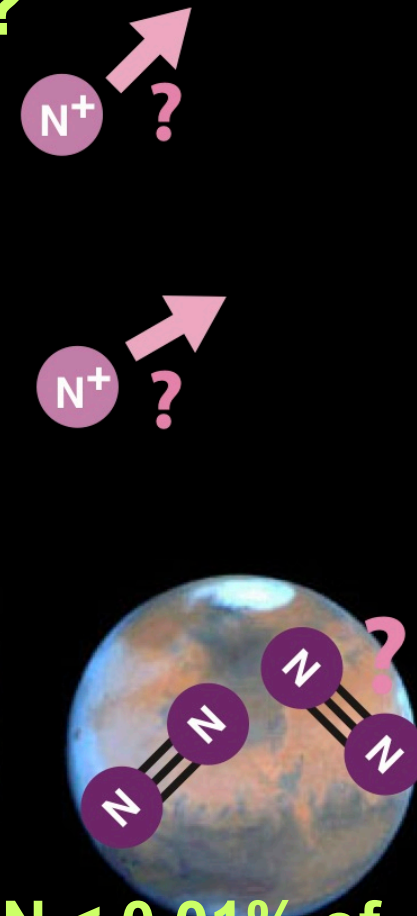
<< @ Earth, Venus, Titan



Venus

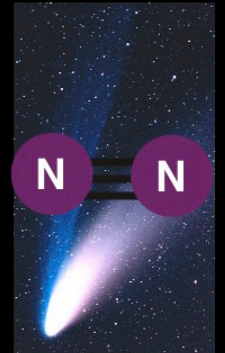


Earth

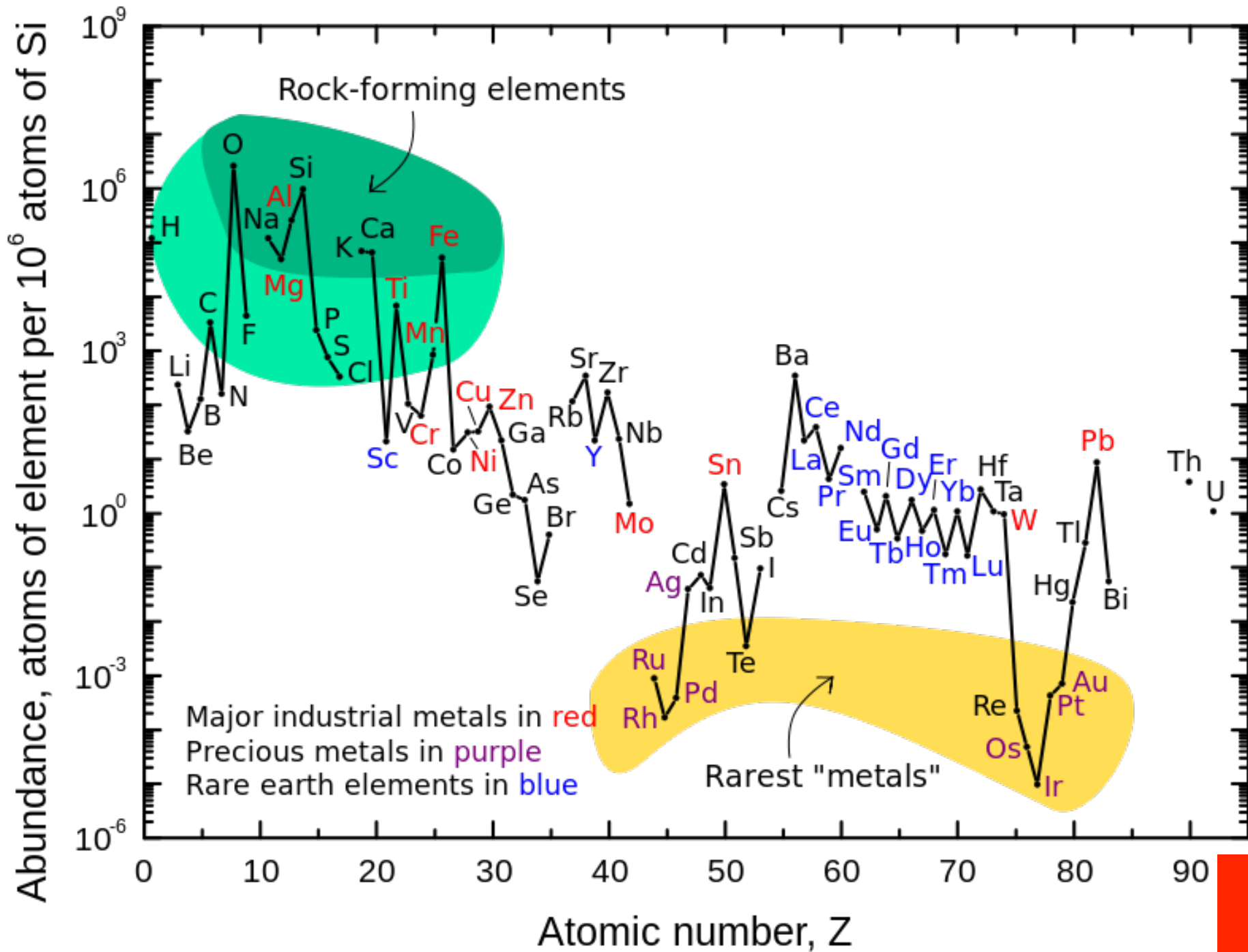


N < 0.01% of Earth/Venus

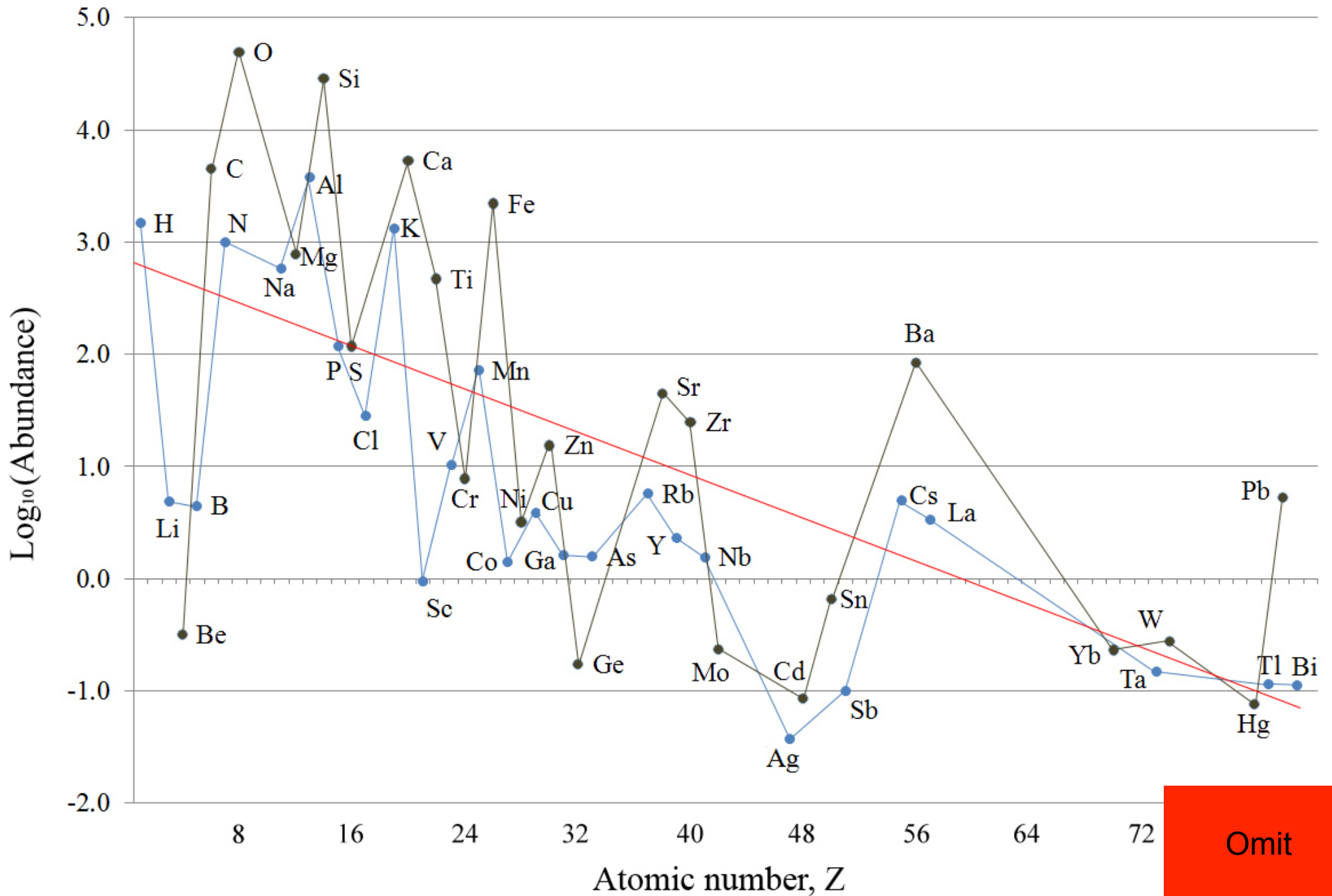
Mars



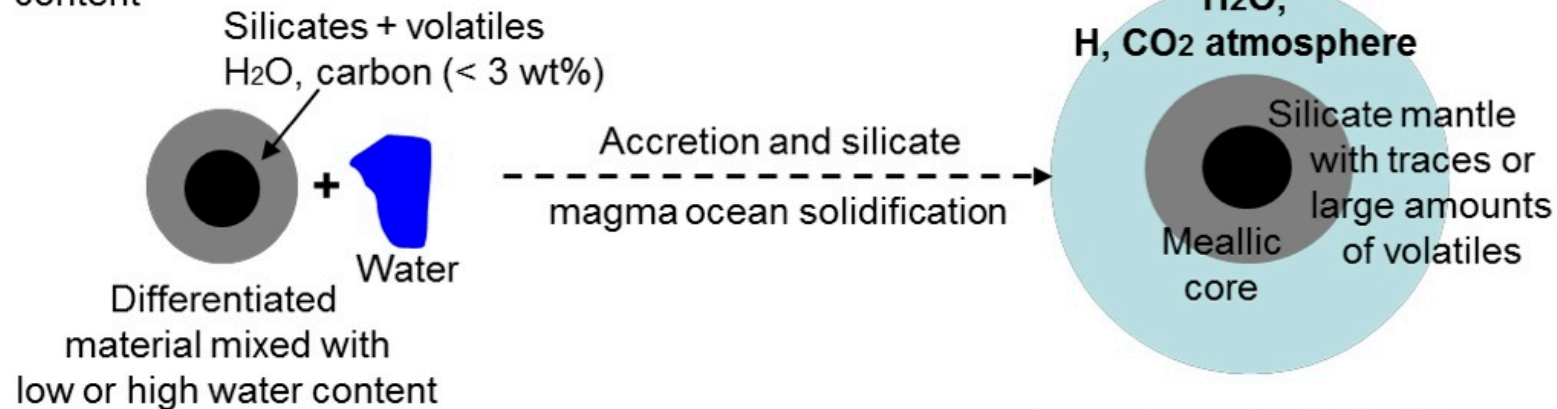
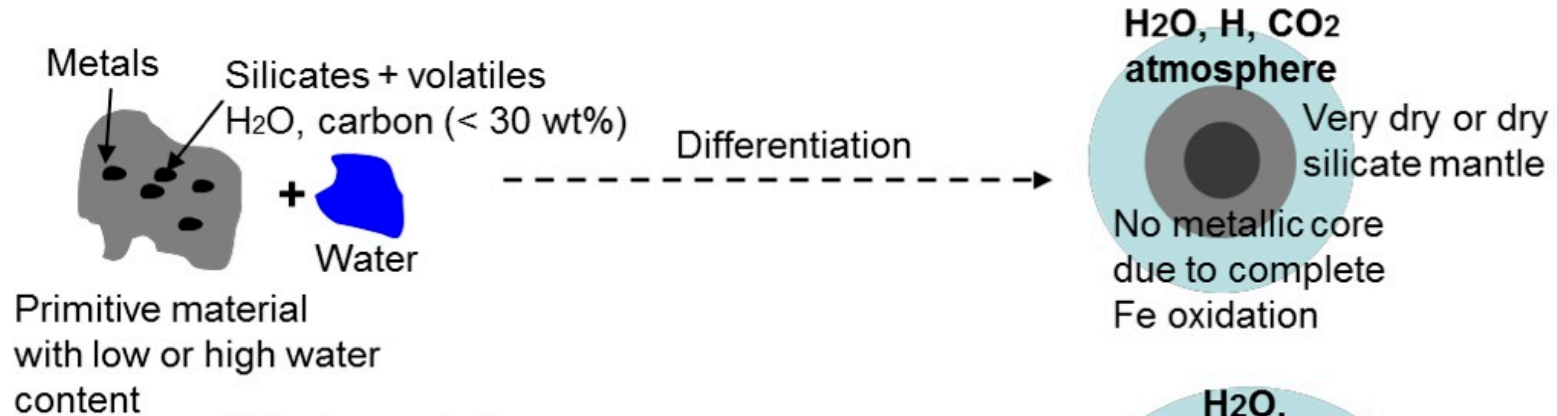
Omit



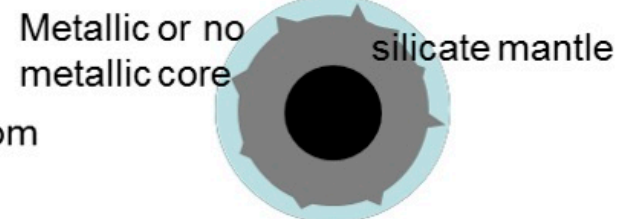
Omit



Atmosphere formation models



Less dense H₂O, H, CO₂ atmosphere



Secondary atmosphere formation due to degassing from subsequent tectonic activity

Omit